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# Experimental modal analysis and finite element model updating for structural health monitoring of reinforced concrete radioactive waste packages

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## Abstract

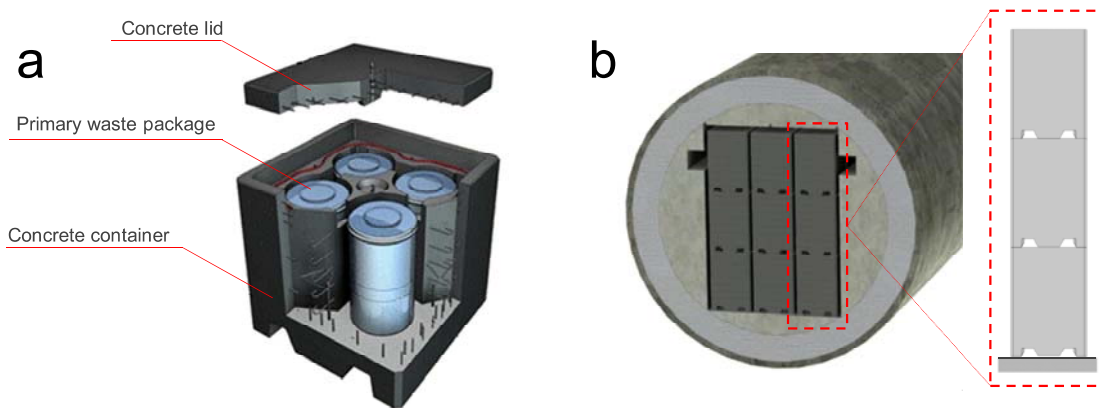
This study envisages the use of modal analysis for monitoring the structural health of radioactive waste packages. To this end, the calibration of a numerical model that describes the dynamic behavior is a critical issue for the success in damage detection. In this study, experimental modal analysis was conducted on a radioactive waste package mockup. The container was tested under different boundary conditions. Then, the experimental modal analysis data was used to update finite element models that describe the observed behaviors. The latter consists in the formulation of an optimization problem that minimizes the differences between the experimental and the numerical data. A two-step methodology is proposed for finite element model updating. First, a full factorial design of experiments allowed estimation of a set of parameters of the numerical model that minimize a cost function. Second, a genetic algorithm was conducted, wherein the initial population of parameters was generated as a function of that set of parameters obtained in the previous step. This study serves as preliminary step towards the implementation of a structural health monitoring based on modal analysis. Specific aspects for the implementation of a modal-based structural health monitoring system in a radioactive waste repository are also summarized.

## Keywords

Finite element model updating, experimental modal analysis, radioactive waste package, reinforced concrete, dynamic properties.

33 **1 Introduction**

34 Andra (the French National Radioactive Waste Management Agency) is designing a  
35 geological repository —project *Cigéo*: “Industrial Center for Geological Disposal”—, for  
36 disposal of long-lived high- and intermediate-level radioactive wastes. Detailed description  
37 of the *Cigéo* project can be consulted in [1]. In the particular case of long-lived  
38 intermediate-level waste, the use of reinforced concrete containers as a disposal package  
39 seems to be the preferred solution in many countries [2], [3]. The waste packages consist of  
40 i) a primary waste package in the form of metallic drums which contain the immobilized  
41 radioactive waste, and ii) a precast reinforced concrete container, which is designed to host  
42 the primary waste package. Reinforced concrete containers provide a physical protection to  
43 the primary waste package, and they can be produced in a shape that eases handling  
44 operations. Figure 1a shows a schematic of a radioactive waste package prototype and its  
45 constitutive parts. This type of waste package prototype is being considered in the *Cigéo*  
46 project. It is foreseen that the waste packages be stacked (up to three packages) and stored  
47 in an underground cell with sufficient capacity. Figure 1b shows a schematic description of  
48 the radioactive waste packages stored in an underground repository cell.



*Figure 1. a) Typical reinforced concrete container designed to host intermediate and low level radioactive waste, and b) schematic description of the final disposition of waste packages in an underground repository cell, and detail of three stacked containers.*

50           The management and control of concrete in radioactive waste repositories entails  
51 several challenges. Despite concrete is being considered for radioactive waste disposal,  
52 there is no previous experience on concrete withstanding nuclear environments beyond its  
53 expected service life; ~100 years and at high temperatures (~65°C), for intermediate-level  
54 waste. A number of studies claim that further research on concrete durability is needed, to  
55 account for the simultaneous and synergistic actions of moisture, thermal, and mechanical  
56 loads along with radiation damage [3]–[8]. In addition, most of these studies also appeal to  
57 the need of developing nondestructive evaluation techniques and structural health  
58 monitoring (SHM) systems, that inform about the mechanical and durability performance  
59 of concrete in nuclear facilities. In particular, the difficulties arise from the limits on access  
60 to the structures and the harsh environment [3]. Previous studies investigated the feasibility  
61 of different nondestructive techniques, for the inspection of concrete in radioactive waste  
62 repositories. Iliopoulos and coworkers [9] combined the application of different  
63 nondestructive techniques on a radioactive waste container prototype subjected to high  
64 temperatures. Digital image correlation and acoustic emission were able to monitor the  
65 evolution of thermal induced cracking damage, while the depth of already formed cracks  
66 was successfully investigated through the ultrasonic pulse velocity. Davis and coworkers  
67 [10] applied different stress wave techniques aiming at periodic inspection of in-service  
68 radioactive waste tanks made of reinforced concrete. The investigated techniques included  
69 impulse response technique, ultrasonic pulse velocity and sonic logging. From this set of  
70 techniques, the authors were able to investigate the uniformity of concrete, and thus detect  
71 vulnerable zones. Andrade and coworkers [5], and Duffó and coworkers [6] have focused  
72 their researches in the application of electrical resistivity measurements, aiming at the  
73 service-life prediction of the steel reinforced concrete containers. In the particular case of  
74 the *Cigéo* project, Andra has instrumented prototype concrete containers with embedded  
75 optical fibers, so mechanical strains can be monitored since their production, and during  
76 their final disposition in the geological repository [11], [12].

77           This study envisages the use of modal analysis for monitoring the structural health  
78 of radioactive waste packages. In modal-based SHM applications, sensors are permanently  
79 installed in structures and continuously record their dynamic characteristics. Since dynamic  
80 properties are related to the mechanical integrity of structures, the eventual apparition of



81 distress may be detected [13]. In turn, the modification of the dynamic properties because  
82 of damage may also provide an indication of its severity and position within the structure.  
83 These contentions lay on the classification of damage assessment methodologies  
84 established by Rytter [14]: i) detect, ii) localize, iii) quantify, and lastly, iv) make a  
85 prognosis of remaining service life. Different approaches for detection and localization of  
86 damage from vibration responses have been reviewed in [15]–[17]. These approaches are  
87 classified as i) response-based methods, or ii) model-based methods. The former only  
88 depends on measured data, and is commonly used for damage identification, and eventually  
89 localization. The latter leverages an analytical or numerical model of the structure at the  
90 intact state, which is used to identify, localize, and in addition quantify the damage severity,  
91 by comparing an updated model at the damaged state [18]. Both approaches (response-  
92 based and model-based) are complimentary, since in most practical cases, response-based  
93 methods are used to detect damage occurrence, while model-based damage are used to  
94 detect and quantify damage severity [18]. Furthermore, the success of model-based  
95 methods depends on the model quality. Very often, model assumptions, and errors in  
96 estimated model parameters (e.g. geometry, material properties, and boundary conditions)  
97 lead to significant discrepancy with regard to experimental data. In this respect, model  
98 updating techniques have become a subject of intense research. Model updating consists in  
99 the process of correcting the relative mismatch between experimental and numerical  
100 modelling data of a structure, for obtaining better agreement between both, and so  
101 improving the predictions of its dynamic and static mechanical behavior [19], [20]. This  
102 implies the formulation of an optimization problem wherein the differences between  
103 experimental and modelling data are minimized.

104           In civil engineering applications, modal analysis and finite element model (FEM)  
105 updating techniques have been applied and adapted for the evaluation and damage  
106 assessment of already existing structures [21], such as bridges [22], [23], footbridges [24],  
107 [25], buildings [25-27], dams [21], towers [28], [29], and cultural heritage [29], [30]. In this  
108 study, experimental modal analysis (EMA) and FEM updating was conducted on a  
109 radioactive waste container prototype. It was tested under different boundary conditions.  
110 First the radioactive waste package was tested empty. Second, after loading it with a  
111 dummy (non-radioactive) primary waste package. Such a dead load (the load of the primary

112 waste package) affected the resonant frequencies of the concrete container. Then, different  
113 finite element models that describe the loaded and unloaded concrete containers are  
114 considered. The parameters of the finite element models are then adjusted to fit the EMA  
115 data. Different algorithms and methodologies have been proposed to solve the optimization  
116 problem [19], [31], being the nonlinear simplex algorithm [32], the trust-region algorithm  
117 [27], neural networks [26], [28], [33], [34] and genetic algorithms [35] the most appealed  
118 alternatives. In this study, a methodology for FEM updating is proposed, and applied on a  
119 radioactive waste package. To do so, an initial global sensitivity analysis is conducted,  
120 which is based on a full factorial design of experiments (DOE). This initial analysis allows  
121 determination of the most influential FEM parameters on the cost function values, and  
122 obtaining indicative values that minimize it. This indicative minimum solution is used to  
123 narrow the space solution for solving the optimization problem through a genetic algorithm.  
124 Finally, a local sensitivity analysis is conducted using a one-at-a-time approach. This  
125 process analyzes the impact of the variations away from the attained optimum value for  
126 every updating parameter. The updated numerical models may serve i) to predict the  
127 mechanical performance of the waste packages under different loading scenarios, which  
128 saves efforts and budget in cumbersome experiments, and ii) to obtain finite element  
129 models for model-based detection techniques, as a baseline condition. To the authors'  
130 knowledge, this is the first study that proposes modal analysis as an alternative to monitor  
131 the mechanical performance of radioactive waste packages. This study constitutes a  
132 preliminary analysis towards the implementation of a modal-based SHM system for  
133 controlling the mechanical integrity of radioactive waste concrete containers. The  
134 implementation of a modal-based SHM on a radioactive waste repository faces particular  
135 challenges which are also summarized herein. These concerns are subject of ongoing  
136 research.

137 In the following, the present article is structured as follows: Section 2 describes the  
138 waste package prototype, and provides a detailed description of the EMA test  
139 configuration. Section 3 describes the experimental results: the obtained resonant frequency  
140 peaks and their corresponding modal shapes. Section 4 presents different finite element  
141 models that describe the experimental data. Section 5 describes the optimization  
142 methodology for finite element model updating, and analyzes the sensitivity of the finite

143 element models to the input parameters. Finally, section 6 summarizes the conclusions that  
 144 are drawn from this study, and summarizes practical aspects and challenges related to the  
 145 implementation of a modal-based SHM strategy on a radioactive waste repository.

## 146 2 Experimental details

### 147 2.1 Description of the radioactive waste package

148 Figure 2a shows a schematic of the waste package and its parts: i) the primary waste  
 149 package in the form of a metallic drum, and ii) a concrete container which is designed to  
 150 host the primary waste package. A lid, also made of reinforced concrete, is fixed to the  
 151 container through four partially threaded screws. The Figure 2b shows the container  
 152 dimensions. The container rests on four L-shaped feet (see Figure 2b). Overall, it measures  
 153 2.25m x 1.54m x 1.54m, and weights roughly 12 tons when it is loaded with the primary  
 154 waste package. Table 1 lists indicative values of mass for the different parts.

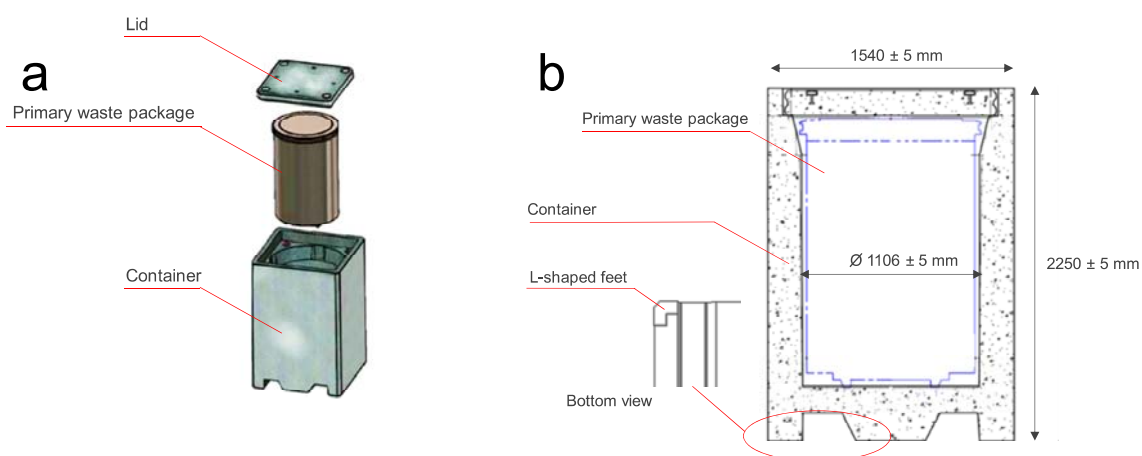


Figure 2. Description of the radioactive waste container prototype: a) schematic of the different parts of the radioactive waste package, and b) detailed dimensions of the concrete container.

155 Table 1. Approximated mass specifications of the different elements of the reinforced concrete  
 156 container prototype.

<b>Total steel reinforcement</b>	360 kg
<b>Lid</b>	662 kg
<b>Primary waste package</b>	3831 kg
<b>Reinforced concrete container (empty)</b>	8417 kg

157 Experimental modal analysis was conducted on a radioactive waste package  
158 prototype. It was tested under two different states. First, the concrete container was tested  
159 empty. At this state the four screws on the lid were incompletely threaded, and a thin foam  
160 was placed in between the two bodies. Second, the concrete container hosted a dummy  
161 (non-radioactive) primary waste package; at this state the four screws on the lid were  
162 completely threaded. The applied torque onto the screws was not ascertained at both states;  
163 yet, they were noticeably different. Figure 3 shows three photos that describe handling  
164 operations of the dummy primary waste package and the concrete lid in between modal  
165 tests. The package was placed on a metallic platform, which was conceived to distribute the  
166 load during mechanical tests. Typical mechanical tests conducted in this platform consist in  
167 piling up several containers, which reproduce the in-service conditions [12]. These tests are  
168 not discussed herein.



*Figure 3. General overview and handling of the dummy primary waste package, a) remove the lid, b) loading the container with a dummy primary waste package, and c) closing of the waste package*

169 Indicative values for elastic moduli and density of the precast concrete container  
170 were obtained on three cylindrical samples (length  $\sim 0.225\text{m}$  and diameter  $\sim 0.115\text{m}$ ), which  
171 were produced with identic concrete composition to the waste packages. To that end, the  
172 dynamic modulus and Poisson's ratio were ascertained according to the standard resonant  
173 frequency method ASTM C215-14 [36]. The dynamic modulus was derived from the  
174 longitudinal frequency mode, and Poisson's ratio was derived from the torsional mode of

175 vibration, using the formulae provided in ASTM C215-14. The density of the concrete  
 176 samples was measured from the mass at the moment of test and the actual dimensions of  
 177 every sample, which were verified with a caliper. At the moment of test, the samples were  
 178 fully matured (more than 1 year after casting). For reference, the 28-day compressive  
 179 strength is ~45 MPa. The Table 2 lists the physical properties of the concrete samples.  
 180 These physical properties are used later on as guidance in the numerical simulations.

181 *Table 2. Physical properties of the concrete samples. Mean values  $\pm$  standard deviation.*

<b>Dynamic modulus, <math>E</math> (GPa)</b>	<b>Poisson's ratio, <math>\nu</math></b>	<b>Density <math>\rho</math> (kg/m<sup>3</sup>)</b>
39 $\pm$ 1	0.18 $\pm$ 0.01	2320 $\pm$ 10

182

## 183 *2.2 Experimental modal analysis*

184 The ways to conduct experimental modal analysis (EMA) are manifold. In this  
 185 study, a roving hammer test was conducted to identify resonant frequencies and modal  
 186 shapes of the concrete container. The advantages of a roving hammer test against other  
 187 alternatives are that it is relatively inexpensive, it can be set-up in very short time, and  
 188 virtually, an unlimited number of degrees of freedom can be tested. The roving hammer test  
 189 was conducted on a regular grid placed on a face of the container. It consisted of 80 degrees  
 190 of freedom: 10 rows by 8 columns every 20cm. The vibration responses were obtained by  
 191 striking the surface of the container with a modal hammer (Bruel & Kjaer model 8207,  
 192 0.225 mV/N). An accelerometer (Bruel & Kjaer model 4525B, 1.046 mV/m·s<sup>-2</sup>) was glued  
 193 on the surface which sensed the out-of-plane vibration. Figure 4 shows a schematic  
 194 representation of the test configuration, and the relative positions of the grid with regard the  
 195 accelerometer.

196 The modal characteristics were extracted using the peak amplitude method [37]. At  
 197 every degree of freedom ( $p$ ), the accelerance  $H_p(\omega)$  is obtained as

$$198 \quad H_p(\omega) = \frac{X_p(\omega)}{F_p(\omega)}, \quad (1)$$

199 where  $F_p(\omega)$  is the Fourier transform of the force signal, and  $X_p(\omega)$  is the Fourier transform  
 200 of the acceleration signal. Since not all resonant frequencies have meaningful spectral  
 201 amplitude at all measured degrees of freedom  $p$ , the squared singular values of the vector  
 202  $[H_1(\omega_i), H_2(\omega_i), \dots, H_{80}(\omega_i)]$ , were computed at every frequency (or spectral line,  $\omega_i$ ), to  
 203 obtain a representative spectrum of the whole grid. This approach is termed complex mode  
 204 indication function [37], [38]. The imaginary part of the accelerance (Eq. 1) is much more  
 205 discriminating with respect to close modal frequencies [38]. Therefore, only the imaginary  
 206 part of the accelerance was used to identify the resonant frequency peaks. Furthermore, the  
 207 corresponding modal shapes were obtained by mapping the corresponding imaginary part  
 208 of Eq. 1 onto the grid. Recall that the imaginary part of the accelerance is proportional to  
 209 the modal displacement at a given resonant frequency peak [37], [39]. Only modes up to  
 210  $\sim 1000$  Hz were considered. This cutoff frequency was determined for a decrease of 90% of  
 211 the input energy force.

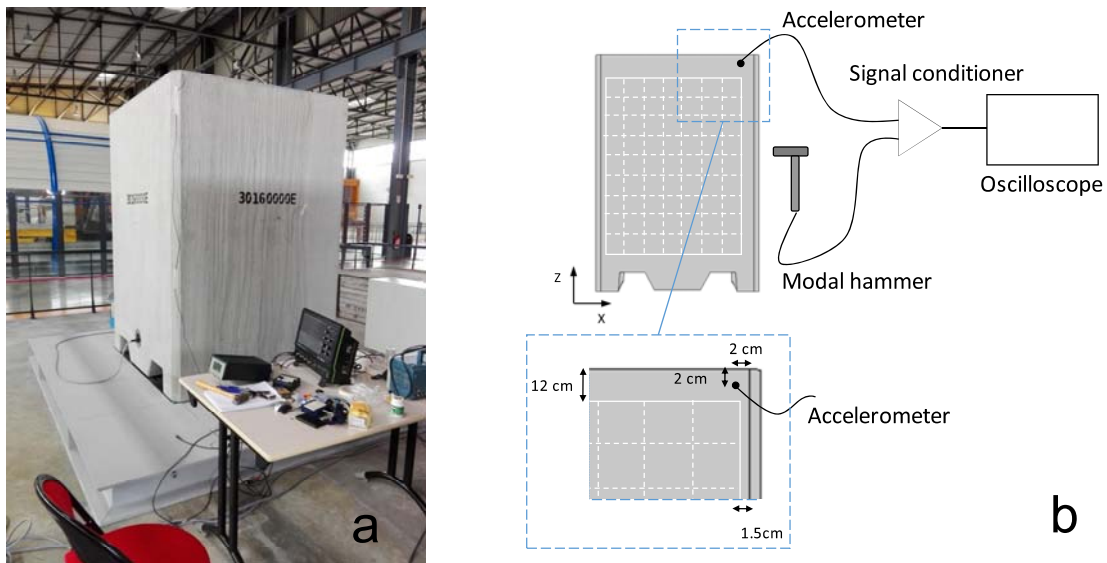
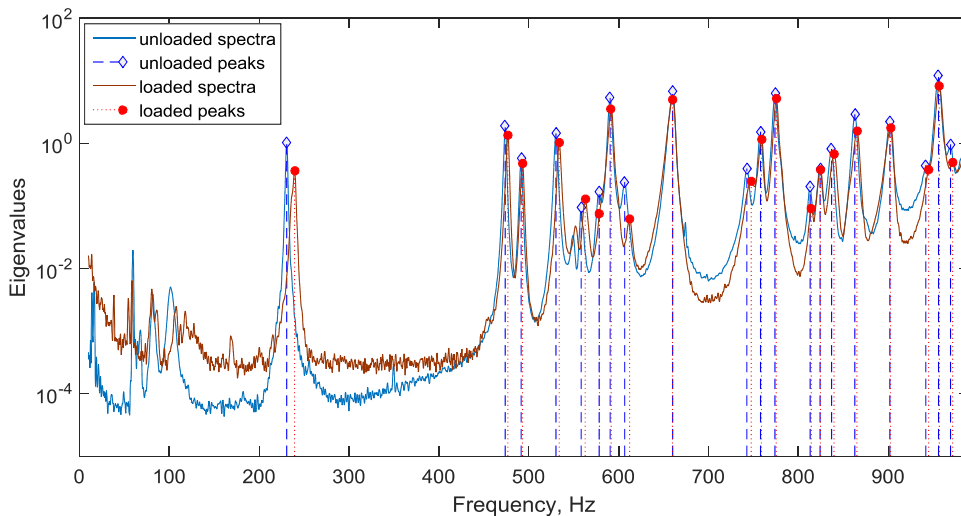


Figure 4. General overview of the container which lays on a metallic platform, and b) schematic representation of the test configuration showing the relative position of the measured degrees of freedom with regard the geometry of the container: a 10x8 grid every 20 cm placed on the front face (not to scale).

### 212 3 Experimental modal analysis results

213 The Figure 5 shows the eigenvalue spectra resulted from the singular value  
 214 decomposition of the imaginary part of the accelerance obtained for the unloaded and

215 loaded conditions. The spectra show different frequency peaks, which frequently appear  
216 closely spaced. In both cases, a number of 20 resonant frequency peaks were identified,  
217 ranging from 200 to 1000 Hz. These are shown as vertical dashed lines in Figure 5. Low  
218 frequency contributions (<100 Hz) were also identified, which correspond to rigid body  
219 modes. However, their amplitudes are two orders of magnitude lower than those  
220 corresponding to resonant mode peaks (note logarithmic scale in Figure 5). Figure 6 shows  
221 the mode shapes corresponding to the 20 extracted resonant frequency peaks at the  
222 unloaded state. Similar mode shapes and in the same order were obtained after loading.  
223 However, after loading, all resonant frequency peaks shifted upward in different extents:  
224 between 0.1 and 4%. The biggest difference was found for the lowest identified resonant  
225 frequency mode. This effect indicates that the load of the dummy primary waste package  
226 modified the dynamic properties of the structure. In the following, finite element modeling  
227 was used to interpret the observed behavior.



*Figure 5. Resonant spectra obtained before and after loading the container with a dummy primary waste package. A number of 20 resonant frequency peaks were detected between 200 Hz and 1000 Hz.*



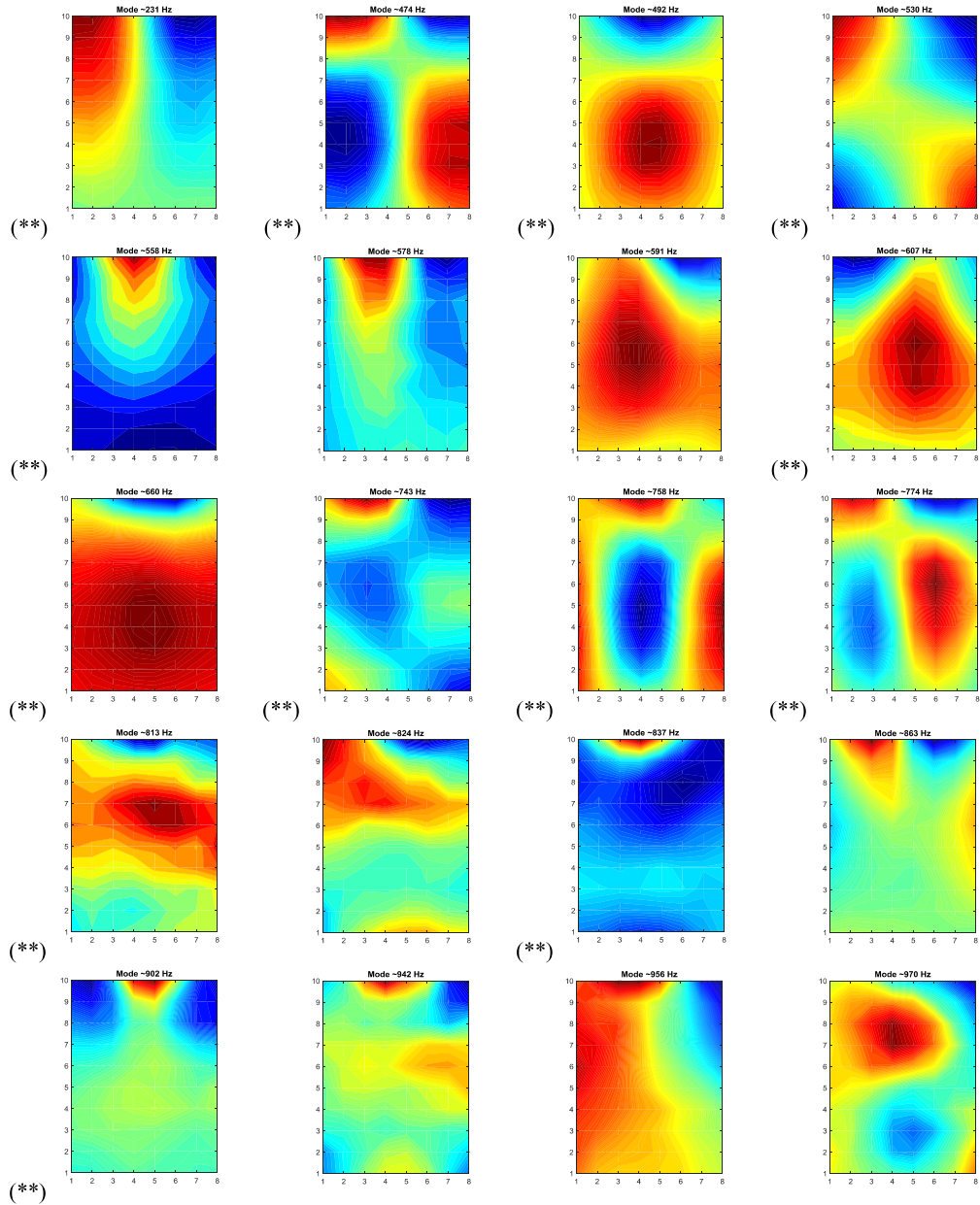


Figure 6. Experimental modal shapes (out-of-plane motion) obtained on the front face of the container. The symbols (\*\*\*) denote that these mode shapes were paired and used for the FEM updating later on.

#### 229 4 Finite element modelling

230 Figures 7a and 7b show a schematic description of the boundary conditions of the  
 231 numerical model of the concrete container, for the unloaded (Figure 7a) and loaded states  
 232 (Figure 7b). Massless spring elements (denoted with subscript *tel*, which stands for thin  
 233 elastic layer) were used to describe the boundary condition between the lid and the concrete



234 container. These springs accommodate the different nodal displacements between the two  
235 bodies in  $x$ ,  $y$  and  $z$  directions, with a preset spring constant ( $K_{tel}$ ). These springs were  
236 considered to have similar stiffness values in every direction, thus  $K_{tel,x} = K_{tel,y} = K_{tel,z}$ .  
237 Analogously, massless spring elements ( $K_{spf}$ , subscript  $spf$  stands for spring foundation)  
238 model the boundary condition between the floor and the four L-shaped feet of the  
239 container. Both,  $K_{spf}$  and  $K_{tel}$  elements were distributed on their respective contacting areas.  
240 Moreover, additional mass ( $A_{mass}$ ) was added to the bodies to consider the mass of the steel  
241 reinforcement, along with deviations from the measured concrete density ( $\rho = 2320 \text{ kg/m}^3$ ).  
242 Therefore, the density of concrete was kept constant in the numerical simulations, and the  
243 parameter  $A_{mass}$  was used as updating parameter. Linear elastic and isotropic behavior was  
244 considered. In the particular case of the loaded container (Figure 7b) the load corresponding  
245 to the weight of the primary waste package was distributed on the available surface ( $\sim 1 \text{ m}^2$ ),  
246 and massless spring elements ( $K_{spf}^l$ ) were used to constrain the movement at this zone.  
247 Again, these springs were considered to have similar stiffness values in  $x$ ,  $y$  and  $z$   
248 directions.

249 Different alternatives were considered to render an explanation of the relative  
250 increase of resonant frequencies with regard the unloaded state. On the one hand, dead  
251 loads on the container may lead to a local increase of stiffness because of geometric  
252 nonlinearity, and so, leading to an increase of resonant frequencies [40], [41]. On the other  
253 hand, nonlinear elastic behavior can further lead to stress dependent resonant frequency  
254 variations [42]. However, for given boundary conditions ( $K_{tel} = K_{spf} = K_{spf}^l = 1 \text{ GN/m/m}^2$ ), and  
255 considering third-order elastic constants for concrete —for instance, those obtained for  
256 concrete by Payan and coworkers [43] or by Lundqvist and Rydén [42]—, the sole effect of  
257 the dead load lead to very weak variations (less than 0.0001%) of resonant frequencies with  
258 regard the same model without load. For reference, the effect of the load of the primary  
259 waste package ( $\sim 3831 \text{ kg}$ ) distributed on the available surface ( $\sim 1 \text{ m}^2$ , see Figure 7b) leads to  
260 a maximum Von Mises stress of roughly  $\sim 0.7 \text{ MPa}$  in the volume of the container. Hence,  
261 the load of the primary waste package container is low enough to disregard nonlinear  
262 behavior. Conversely, the sole incorporation of the spring elements that constrain the  
263 loaded zone ( $K_{spf}^l$ ) cause a meaningful increase of the resonant frequencies with regard the  
264 unloaded model (Figure 7a). In addition, the dead load may plausibly further constrain the

265 boundary conditions on the four feet, and so, contributing also to increase the resonant  
 266 frequencies. This latter option was considered henceforth.

267 A number of  $N= 45$  eigen-frequencies ( $f_n$ ) and their corresponding modal shapes  
 268 ( $\phi_n$ ) were computed numerically for both models (Figures 7a and 7b). Figure 7c shows a set  
 269 of representative mode shapes extracted out of the 45 computed modes. They were obtained  
 270 for the unloaded state, and for the parameter values of  $K_{tel} = K_{spf} = 1 \text{ GN/m/m}^2$ ,  $A_{mass} = 360$   
 271 kg, and the material properties listed in Table 2. A number of modal shapes correspond to  
 272 the own resonant frequencies of the lid. For this particular set of model parameters, the  
 273 modes 1 to 12, 15 to 17, 26, 27 and 45 correspond to the own resonant frequencies of the  
 274 lid; an example of a resonant frequency of the lid is shown in the last picture in Figure 7c.  
 275 Experimental and numerical mode shapes were compared through the Modal Assurance  
 276 Criterion ( $MAC$ ) as [44],

$$277 \quad MAC_{m,n} = \frac{|\phi_n \cdot \phi'_m|^2}{|\phi_n \cdot \phi'_n| \cdot |\phi_m \cdot \phi'_m|}, \quad (2)$$

278 wherein  $\phi_m$  is the modal shape of the  $m$  experimental mode, and  $\phi_n$  is the numerical modal  
 279 shape corresponding to the  $n$  eigen-frequency. The  $MAC$  is a measure of the similarity  
 280 between the  $m$  experimental modal shape, and the  $n$  numerical modal shape.  $MAC$  values  
 281 vary between 0 and 1. Figure 8 shows the pairwise comparison of the 20 extracted  
 282 experimental modal shapes (those shown in Figure 6) with the 45 computed numerical  
 283 modal shapes (those shown in Figure 7c). Herein, a pair was assigned for the maximum  
 284 attained  $MAC$  value, and if that was higher than 0.65. Moreover, those resonant modes  
 285 corresponding to the lid were disregarded, since they produce negligible out-of-plane  
 286 displacement on the measured surface. In some cases, the experimental modal shapes could  
 287 be attributed to several numerical modal shapes. For instance, the 16<sup>th</sup> experimental mode  
 288 shape could be assigned to two different numerical modes (see Figure 8, and inset  
 289 captions). This particular mode of vibration was not used to prevent from false assignation  
 290 during the optimization process. As a result, a number of 13 modes could unequivocally be  
 291 assigned, out of the 45 numerical modes (circle markers in Figure 8). These vibration  
 292 modes were retained for the inversion of the parameters of the numerical model.

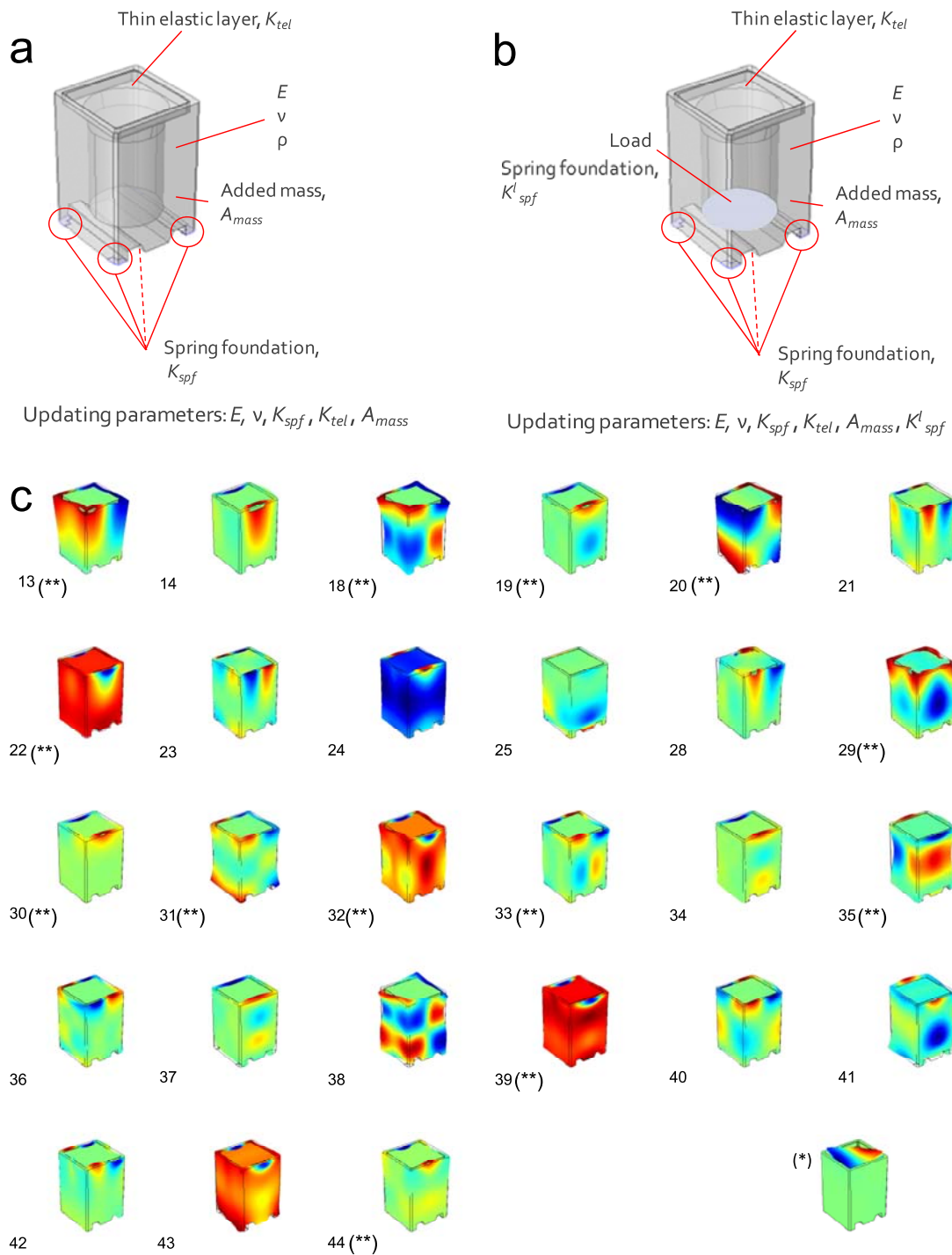


Figure 7. Description of the boundary conditions and the physical parameters of the numerical mode, a) unloaded, and b) loaded container. c) Representative modal shapes corresponding to the out-of-plane displacement of the front face of the unloaded container; (\*) last picture (bottom right) shows a representative mode of vibration of the lid. The symbols (\*\*) denote that these mode shapes were paired and used for the FEM updating.

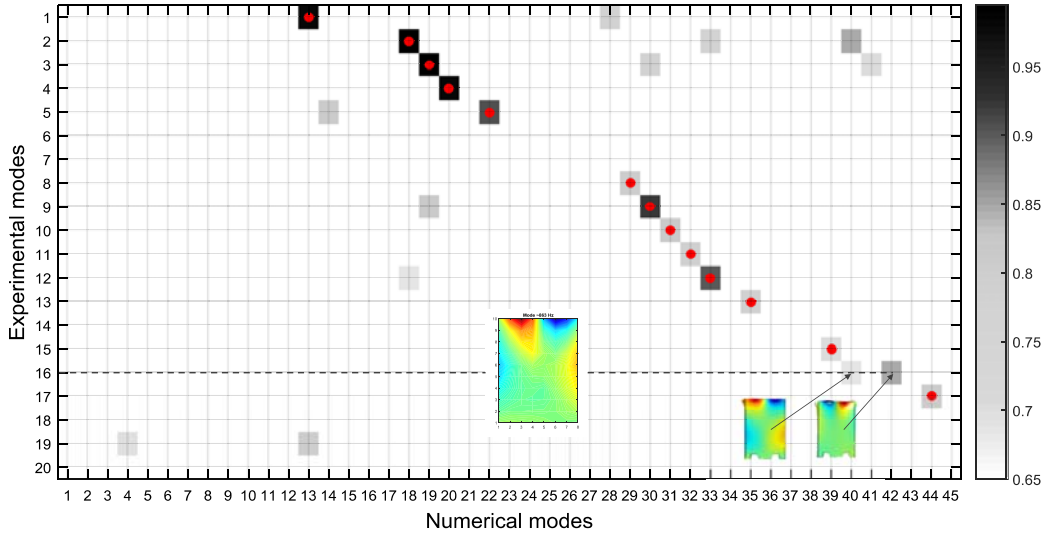


Figure 8. Modal Assurance Criterion values obtained for pairwise comparisons between numerical and experimental modal shapes. Circle markers represent paired modes, which were retained for the optimization problem.

293

## 294 5 Optimization problem and sensitivity analysis

295 Given that the actual material properties and boundary conditions are unknown, an  
 296 optimization problem can be formulated. That is, to find the values of the set of unknown  
 297 parameters of the numerical model ( $\bar{p}$ ), that minimize the differences between  
 298 experimental and numerical results. In case of the numerical model corresponding to the  
 299 unloaded state, the set of the unknown parameters are: the modulus ( $E$ ), the Poisson's ratio  
 300 ( $\nu$ ), the spring constant of the spring foundation ( $K_{spf}$ ), the spring constant of the thin elastic  
 301 layer ( $K_{tel}$ ), and the added mass ( $A_{mass}$ ). Therefore, the optimization problem can be  
 302 formulated as

$$303 \quad \min f_c(\bar{p}), \quad (3)$$

304 subjected to upper and lower boundaries,  $\bar{p}_{lower} \leq \bar{p} \leq \bar{p}_{upper}$ , and where the cost function  
 305  $f_c(\bar{p})$  can be written as the sum of weighted squared relative differences between  
 306 experimental (subscript  $m$ ) and numerical (subscript  $n$ ) frequency values as

$$f_c(\bar{p}) = 1/N \cdot \sum_{m=1}^M \sum_{n=1}^N w_{n,m} \cdot \left( \frac{f_n - f_m}{f_m} \right)^2; \quad (4)$$

308  $w_{n,m}$  is a weighting factor that equals 1 if the mode is paired, and 0 elsewhere. Note that not  
 309 all selected experimental modes may be paired during the optimization process, hence the  
 310 cost function in Eq. 4 represents an average error per mode. Other cost functions have been  
 311 considered elsewhere wherein *MAC* values also weight on the cost function; see for  
 312 instance references [23] and [31].

313 As a first approximation to the optimization problem a full factorial three level ( $3^k$ )  
 314 design of experiments (DOE) was conducted;  $k$  stands for the total number of factors,  
 315 herein  $k=5$  for the numerical model corresponding to the unloaded state (see Figure 7a), and  
 316  $k=6$  for the numerical model corresponding to the loaded state (see Figure 7b). The  
 317 objectives of this initial analysis are i) to estimate the global sensitivity of the numerical  
 318 model to model parameters variations, and ii) get an initial estimation of the parameter  
 319 values that minimize the cost function (Eq. 4). This was achieved through the surface  
 320 response methodology [45]. The surface response methodology consists in fitting an  
 321 empirical model to the data generated by the application of a DOE, and find their critical  
 322 values (in this case the minimum). The  $3^k$  full factorial design allows estimation of a  
 323 quadratic model as

$$f_c(\bar{p}) = a_0 + \sum_{i=1}^k a_i X_i + \sum_{i=1}^k \sum_{j=1, j < i}^k a_{ij} X_i X_j + \sum_{i=1}^k a_{ii} X_{ii}^2 + error, \quad (5)$$

325 wherein,  $a_0, a_i, a_j, a_{ii}$  and  $a_{ij}$  are the fitting coefficients of the quadratic model, and the set  
 326 of parameters is  $\bar{p} = [X_1, \dots, X_k]$ . Table 3 shows the factor levels for the set of parameters  
 327 considered in the numerical models: 5 parameters for the unloaded state ( $X_1$  to  $X_5$ ), and 6  
 328 parameters for the loaded state ( $X_1$  to  $X_6$ ). The factor levels were chosen close to the  
 329 indicative values obtained for the elastic modulus and Poisson's ratio of the concrete (those  
 330 listed in Table 2). Conversely, since indicative values for boundary conditions ( $K_{spf}$ ,  $K_{tel}$ ,  
 331 and  $K_{spf}^l$ ) are not available, these were allowed to vary in a wider range. The numerical

332 models were evaluated at the resulting 243 parameter combinations for the unloaded state,  
333 and at the 729 for the loaded state. Then, the resulted numerical eigen-frequencies and  
334 modal shapes were paired to the EMA data, and so the cost function (Eq. 4) was evaluated  
335 for every parameter combination. Once the empirical model in Eq. 5 is fitted to the cost  
336 function values obtained for all the  $3^k$  combinations of parameters, a minimum point can be  
337 estimated. Such estimated minimum can be then used to narrow the space solution to be  
338 explored by a genetic algorithm (GA). From an initial population (say sets of parameters),  
339 genetic algorithms mimic “natural selection” to select sets of parameters that enhance the  
340 cost function. The members of the population are combined and selected as a function of  
341 how they perform on the cost function (or fitness value), to reproduce an enhanced  
342 generation. Crossover and mutation operators are applied through generations. The  
343 crossover function defines how members of the population create next generation, and the  
344 mutation function, introduces diversity in the population by introducing random changes on  
345 them. In general, the initial population can be generated randomly. However, the use of a  
346 priori knowledge may accelerate the convergence to a solution. Herein, the initial  
347 population was narrowed to  $\pm 10\%$  of the estimated minimum by the surface response  
348 method. Note, that the space solution is not strictly constrained within these boundaries, but  
349 they serve to provide an initial population which is randomly generated within these  
350 boundaries. There is a number of applications wherein the application of the surface  
351 response methodology has been successfully combined with genetic algorithms in  
352 optimization problems [46]. The population size was set to 25 in both cases. The crossover  
353 and mutation rates were set to 0.8 and 0.1. The GA was stopped once a preset number of  
354 generations or a threshold value of the cost function is attained; herein, 25 generations or  
355  $10^{-4}$  respectively. Further refinement in the optimization problem can be achieved by using  
356 the obtained solution by the GA algorithm, as starting point in a nonlinear constrained  
357 minimization; herein, through the interior point algorithm. Examples of successful  
358 application of such hybrid genetic algorithms for FEM updating applications can be found  
359 in [23], [31], [34]. Figure 9 summarizes the optimization process conducted herein.

360

361

362 Table 3. Factor levels used in the three level full factorial design.

Parameter		Upper level	Intermediate level	Lower level
X <sub>1</sub>	Modulus (GPa)	40	35	30
X <sub>2</sub>	Poisson's ratio	0.300	0.225	0.150
X <sub>3</sub>	$K_{spf}$ (GN/m/m <sup>2</sup> )	10.00	5.05	0.10
X <sub>4</sub>	$K_{tel}$ (GN/m/m <sup>2</sup> )	10.00	5.05	0.10
X <sub>5</sub>	$A_{mass}$ (Kg)	1000	500	0
X <sub>6</sub>	$K_{spf}$ (GN/m/m <sup>2</sup> )	10.00	5.05	0.10

363

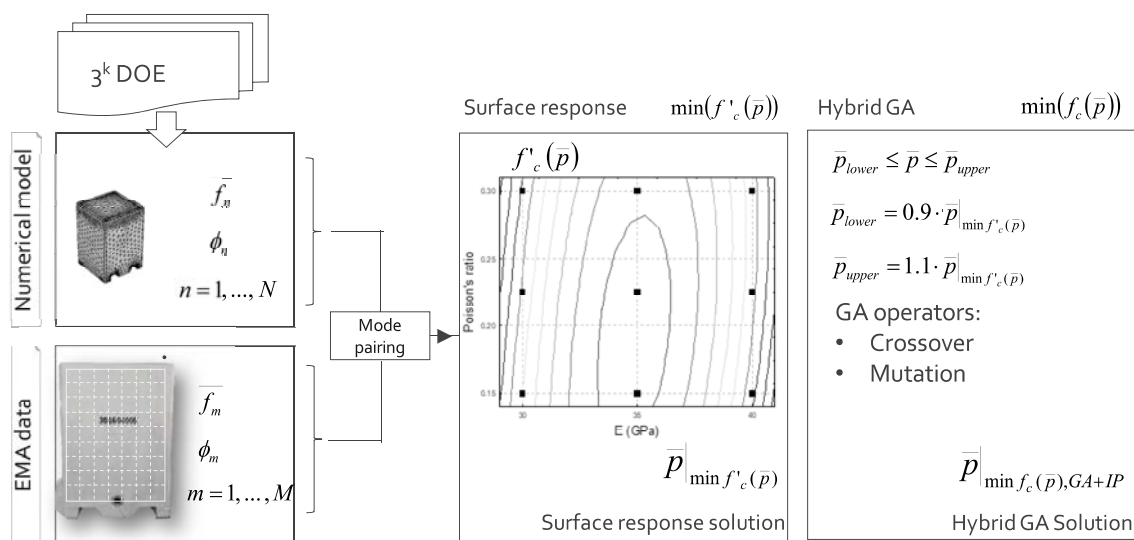


Figure 9. Flowchart representing the data analysis conducted for FEM updating in this study.

364 One of the interests of performing a  $3^k$  DOE is to analyze the global sensitivity of  
 365 the FEM to the model parameter variations. For the sake of conciseness, only the main  
 366 effects are shown. Figures 10a to 10f show the main effects for the cost function. From  
 367 results it is drawn that the cost function is sensible to the variations of  $E$ ,  $K_{tel}$ , and  $A_{mass}$ ,  
 368 within the levels of the DOE. It was the less for all the cases where  $E$  was set to 35 GPa  
 369 (Figure 10a). In addition, the cost function increases monotonically when  $K_{tel}$  increases  
 370 (Figure 10d), and decreased with increasing  $A_{mass}$  (Figure 10e). Conversely, the cost  
 371 function showed less variability with the variations of Poisson's ratio (Figure 10b),  $K_{spf}$   
 372 (Figure 10c), and  $K_{spf}^l$  (Figure 10f). Yet, the regression analysis showed that all the linear  
 373 terms were statistically significant in both cases, including a number of quadratic and the  
 374 interaction terms in the Eq. 5 (we deemed statistical significance for p-values<0.05; the

375 regression results are not shown herein). The explained variance was 98% for the unloaded  
 376 case, and 85% for the loaded one. Subsequently, the fitted equations were used to obtain the  
 377 set of parameters that minimize  $f'_c$ , and initialize the GA.

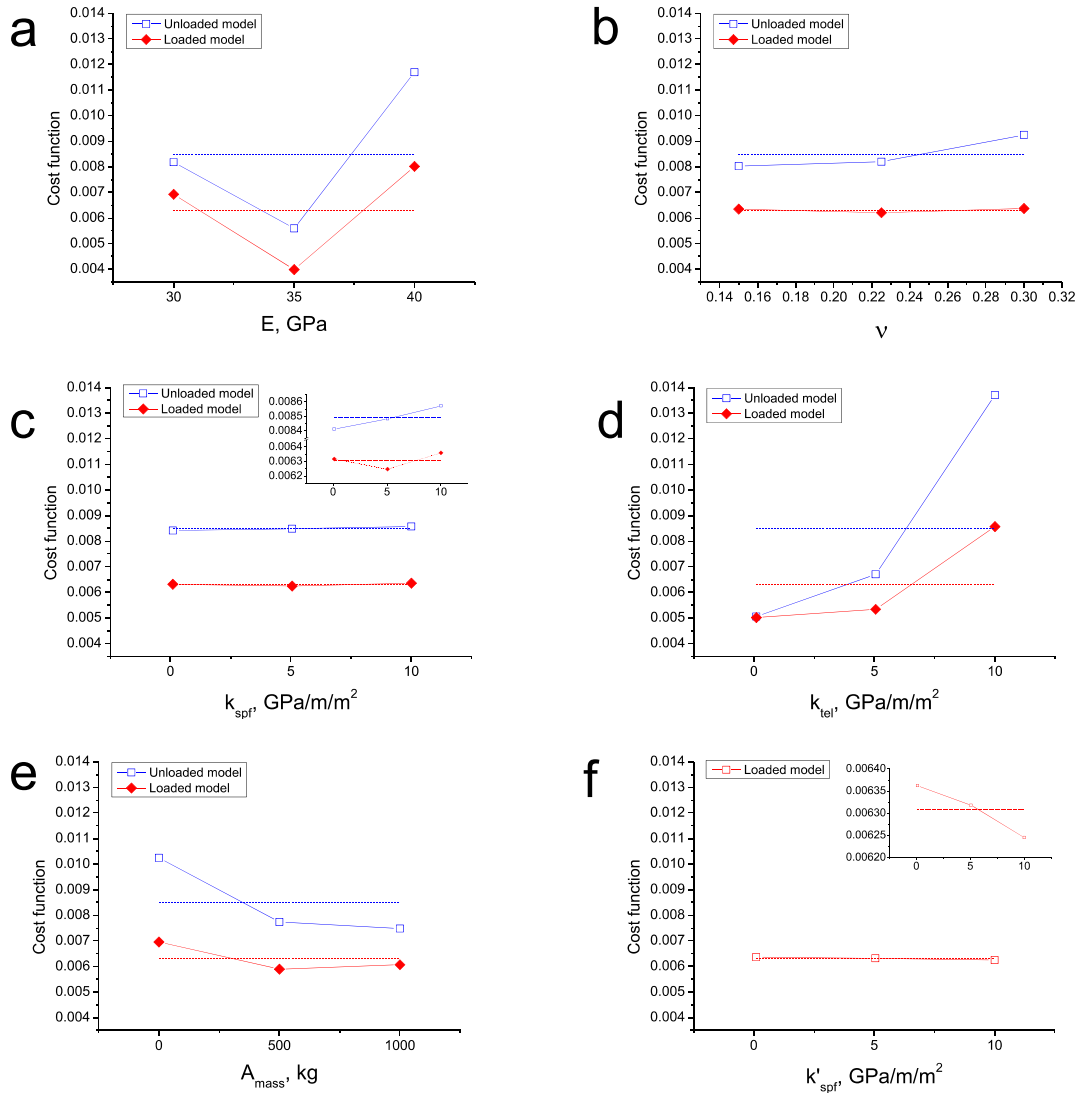


Figure 10. Main effects plots (data means) for cost function values investigated at the  $3^k$  DOE points, for loaded and unloaded FEM. a) Elastic modulus, b) Poisson's ratio, c)  $K_{spf}^1$ , d)  $K_{tel}^1$ , e)  $A_{mass}$ , and f)  $K_{spf}^1$ . Dashed lines represent the global mean.

378 Table 4 summarizes the results of the FEM updating for unloaded and loaded states.  
 379 For reference, intermediate solutions across the different steps of the optimization process  
 380 are also listed: the best  $3^k$  DOE point, and the surface response minimum. The elastic  
 381 properties ( $E$  and  $\nu$ ) were found to be in good agreement with those found for cylindrical



382 samples (Table 2). From results, it is found that the loaded containers may be well  
383 represented by increasing the stiffness of the spring foundation elements. Moreover, Table  
384 5 compares the EMA data and the FEM results obtained during the different steps, and for  
385 both investigated states. *MAC* values remained almost constant across the optimization  
386 steps, while the resonant frequency values varied in larger extent. The symbol (--) denotes  
387 modes that were not paired; thus  $MAC < 0.65$  at this optimization step. Finally, Figure 11a  
388 and 11b show the predicted values as a function of the observed values and the resulted  
389 residuals for both states. Noticeably, the first four resonance modes are predicted with a  
390 relative error below 1%. However, at higher frequencies, the relative error increases,  
391 roughly to 2%. The relative misfits in FEM updating are in general attributed to the model  
392 assumptions: i) model structure errors, ii) model order errors (nonlinear behavior); iii)  
393 model parameter errors, and iv) errors in experimental data [20]. Notwithstanding, the  
394 resulting numerical models describe fairly well the dynamic mechanical behavior of the  
395 packages. These results are expected to be used as a reference condition for future  
396 experimental campaigns. Finally, a local sensitivity analysis was conducted through a one-  
397 at-a-time approach. It consists in investigate the effect of every parameter individually to a  
398 higher and a lower value, while keeping other parameters at the reference value. Herein, we  
399 investigated the local sensitivity within a  $\pm 10\%$  parameter variations. Figures 11c and 11d  
400 show the local sensitivity analysis results for unloaded and loaded states. From, these  
401 results it is drawn that the cost function is dominated by the elastic modulus variations in  
402 both cases, while Poisson's ratio, the added mass, and the boundary conditions variations  
403 lead to minor variations of the result.

404 *Table 4. Results of the set of the parameters of the numerical mode for both investigated states.*

	<b>E</b> (GPa)	<b>v</b>	<b><math>K_{spf}</math></b> (GN/ m/m <sup>2</sup> )	<b><math>K_{tel}</math></b> (GN/ m/m <sup>2</sup> )	<b><math>A_{mass}</math></b> (kg)	<b><math>K_{spf}^d</math></b> (GN/ m/m <sup>2</sup> )	<b><math>f'_c</math></b> ( $\cdot 10^{-4}$ )	<b><math>f_c</math></b> ( $\cdot 10^{-4}$ )
<b>Unloaded (Figure 7a)</b>								
Best 3 <sup>k</sup> DOE point	35.00	0.15	5.05	0.10	500.0	--	--	3.16
Surface response minimum	38.74	0.18	0.10	0.45	967.0	--	1.82	2.04
Hybrid GA optimization	35.85	0.17	5.26	0.77	576.8	--		1.77
<b>Loaded (Figure 7b)</b>								
Best 3 <sup>k</sup> DOE point	35.00	0.30	10.00	0.10	0	5.005	--	5.89
Surface response minimum	38.24	0.29	10.00	1.27	1000	10.00	12.63	3.57
Hybrid GA optimization	35.59	0.19	7.72	1.23	524.7	6.58	--	1.98

405 Table 5. Comparison of EMA data and FEM results across the different optimization steps.

	Mode	EMA	Best 3 <sup>k</sup> DOE point		Surface response minimum		Hybrid GA optimization	
		Freq. (Hz)	Freq. (Hz)	MAC	Freq. (Hz)	MAC	Freq. (Hz)	MAC
Unloaded	1	230.7	220.9	0.98	228.4	0.98	231.1	0.98
	2	474.0	470.9	0.99	479.5	0.99	475.3	0.99
	3	491.7	486.0	0.99	493.9	0.99	489.7	0.99
	4	530.3	521.4	0.98	518.7	0.98	523.1	0.98
	5	558.3	540.9	0.89	548.2	0.90	544.6	0.90
	6	606.9	605.8	0.82	616.5	0.80	612.1	0.81
	7	660.2	659.4	0.92	675.2	0.92	667.9	0.92
	8	742.8	749.2	0.81	740.4	0.82	751.6	0.82
	9	758.0	757.6	0.81	769.6	0.79	763.3	0.80
	10	774.0	761.3	0.90	775.8	0.90	767.1	0.90
	11	813.1	802.5	0.76	818.5	0.77	807.8	0.75
	12	837.1	842.0	0.74	852.2	0.70	846.3	0.74
	13	901.9	--	--	--	--	928.2	0.83
Loaded	1	239.5	232.3	0.99	241.03	0.99	237.7	0.99
	2	476.8	485.0	0.99	480.16	0.99	479.4	0.99
	3	493.2	490.8	0.99	486.20	0.99	491.4	0.99
	4	534.0	524.3	0.98	524.19	0.98	534.3	0.98
	5	562.8	553.0	0.75	550.09	0.75	553.6	0.75
	6	612.3	622.5	0.73	620.08	0.82	617.2	0.74
	7	660.0	686.1	0.95	675.48	0.94	670.6	0.94
	8	747.7	764.8	0.67	752.84	0.67	760.9	0.65
	9	759.1	786.7	0.81	772.25	0.81	766.1	0.80
	10	775.5	790.0	0.89	775.50	0.88	769.4	0.88
	11	824.7	--	--	--	--	--	--
	12	839.7	--	--	839.83	0.83	848.6	0.70
	13	902.8	--	--	944.00	0.89	933.0	0.85

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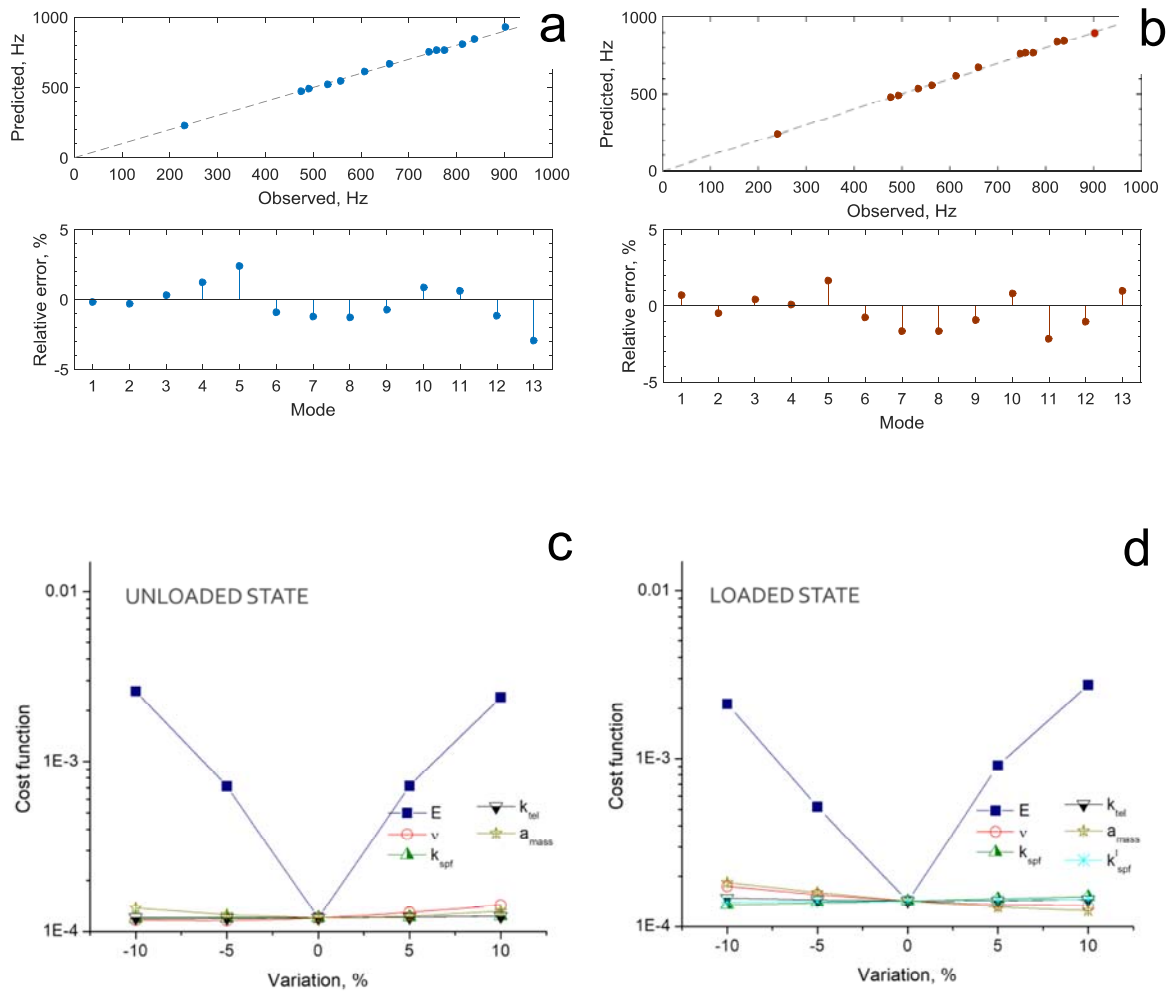


Figure 11. Comparison between experimental and numerical frequencies after the optimization for a) unloaded case, and b) loaded case, and sensitivity analysis, and one-at-a-time analysis for  $\pm 10\%$  variations for c) unloaded, and d) loaded states.

410

## 411 6 Conclusions and prospects

412 Experimental modal analysis has been applied on a prototype radioactive waste  
 413 package. The prototype was tested under two different scenarios: the concrete container  
 414 was tested empty, and then, hosting a dummy primary waste package. Similar modal shapes  
 415 were identified in both cases. However, after loading the concrete container, all resonant  
 416 frequency modes consistently shifted upwards: between 4% and 0.1%. EMA data (resonant

417 frequencies and modal shapes) were used to benchmark finite element models that describe  
418 the dynamic mechanical behavior of the concrete containers at both states. The latter was  
419 achieved through the formulation of an optimization problem that consists in finding the  
420 parameters of the numerical model that minimize the differences with regard the EMA data.  
421 Herein, we first used a full factorial DOE which allowed investigation of the global model  
422 sensitivity to the input parameters. Such an analysis allows screening the parameters that  
423 dominate the dynamic properties within a range of values, get insight of the model quality,  
424 and get an initial estimation of the set of the numerical model parameters that minimize the  
425 differences with respect the EMA data. The optimization problem was subsequently solved  
426 through a genetic algorithm, within a set of parameter boundaries ensuing the analysis of  
427 the DOE. Such a methodology presented in this study was able to successfully calibrate  
428 finite element models that describe the mechanical behavior of the radioactive waste  
429 packages. Numerical models can be then used to predict the mechanical behavior of the  
430 radioactive waste prototypes, which may save considerable labor and budget in  
431 cumbersome experiments, and serve as baseline condition for the implementation of a  
432 modal-based SHM. Model updating can be used also to detect, locate and quantify damage  
433 severity [27], [34]. With regard this latter purpose, the following issues are subject of  
434 ongoing research:

- 435 (i) In the framework of the *Cigéo* project, it is foreseen that the containers be  
436 stacked (see Figure 1b). Thus, three different situations must be  
437 considered depending on the position of the container, each of them,  
438 being subjected to different boundary conditions. A similar FEM  
439 updating procedure can be conducted to obtain numerical models of the  
440 concrete containers at their final disposition in the radioactive waste  
441 repository.
- 442 (ii) The radioactive waste repository is expected to work at temperatures up to  
443  $\sim 65^{\circ}\text{C}$ , depending on the waste type. Previous researches [7], [47] have  
444 demonstrated that at and above such temperatures drying shrinkage in  
445 concrete leads to a decrease of the elastic properties. Furthermore, the  
446 elastic properties of concrete (as other ceramic materials) soften with  
447 increasing temperature [48], [49]. Therefore, eventual temperature

448 fluctuations in the radioactive waste repository along with concrete  
449 aging are expected to cause variations on dynamic properties, and yet not  
450 being indicative of structural damage; or in a reverse sense, obscuring  
451 the presence of damage [16], [50]. These contentions oblige  
452 consideration of additional sensors that monitor operational conditions,  
453 and aid to remove the trends on the modal data driven by such. The  
454 interaction of these effects deserves further study.

455 (iii) In most practical cases, modal-based SHM leverages ambient vibrations  
456 (say wind loads, traffic, or seismic events) to obtain the dynamic  
457 characteristics of structures. Ambient vibrations do not seem feasible in  
458 the case of a geological radioactive waste repository. Therefore, practical  
459 implementation of a modal-based SHM appeals to the need of forced  
460 excitation systems, either shock or harmonic excitation. See for instance  
461 references [21], [51], [52], wherein different excitation devices have  
462 been successfully applied to determine the dynamic characteristics of  
463 buildings. These forced excitation alternatives can be built up in a robot,  
464 or permanently installed in the repository, for promoting the excitation  
465 of resonant frequencies periodically.

466 (iv) Finally, practical implementation of a modal-based SHM system must  
467 consider a cost-effective disposition of sensors regarding number, power  
468 consumption, maintenance and that in turn, maximize the probability of  
469 damage detection [53].

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