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Convection at an isothermal wall in an enclosure and establishment of stratification

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In this experimental-theoretical investigation we consider a turbulent plume generated by an isothermal wall in a closed cavity and the formation of heat stratification in the interior. The buoyancy of the plume near the wall and the temperature stratification are measured across a vertical plane with the temperature laser induced fluorescence method, which is shown to be accurate and efficient (precision of 0.2°C) for experimental studies on convection. The simultaneous measurement of the velocity field with particle image velocimetry allows for the calculation of the flow characteristics such as the Richardson number and Reynolds stress. This enables us to give a refined description of the wall plume, as well as the circulation and evolution of the stratification in the interior. The wall plume is found to have an inner layer close to the heated boundary with a laminar transport of hardly mixed fluid which causes a relatively warm top layer and an outer layer with a transition from laminar to turbulent at a considerable height. The measured entrainment coefficient is found to be dramatically influenced by the increase in stratification of the ambient fluid.

To model the flow, the entrainment model of Morton *et al.* (1956) has first been adapted to the case of an isothermal wall. Differences due to their boundary condition of a constant buoyancy flux, modelled with salt by Cooper & Hunt (2010), turn out to be small. Next, to include the laminar-turbulent transition of the boundary layer, a hybrid model is constructed which is based on the similarity solutions reported by Worster & Leitch (1985) for the laminar part and the entrainment model for the turbulent part. Finally, the observed variation of the global entrainment coefficient, which is due to the increased presence of an upper stratified layer with a relatively low entrainment coefficient, is incorporated into both models. All models show reasonable agreement with experimental measurements for the volume, momentum and buoyancy fluxes as well as for the evolution of the stratification in the interior. In particular the introduction of the variable entrainment coefficient improves all models significantly.

Key words: convection in cavities, plumes/thermals, stratified turbulence

1. Introduction

When a vertical wall is heated above the temperature of the ambient fluid, a boundary layer of hot fluid adjacent to it grows in thickness until it becomes unstable and evolves into a plume-like motion (see e.g. Schlichting & Gersten 2000). When this situation is met in a room, container or reservoir, entrainment and mixing with ambient fluid in

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the closed cavity establishes a stable density stratification (see [Baines & Turner 1969](#)). This stratification limits the vertical circulation and thus plays an important role in heat transfer as well as for the dispersion of tracers. The evolution is therefore of interest to respectively energy saving applications, the quality of water in reservoirs and air circulation in buildings ([Linden 1999](#)).

Here, we consider a heated wall that generates a turbulent boundary layer, hereinafter called the wall plume. This wall plume differs from the classic plume as generated by a point source: since heat is forced over the entire depth, the wall imposes zero velocity and the wall blocks the typical meandering plume motion. The entrainment model of [Morton *et al.* \(1956\)](#) (MTT) has proven very successful for the modelling of plumes and ambient stratification generated by point sources (see [Linden 1999](#); [Kaye 2008](#); [Woods 2010](#); [Hunt & van den Bremer 2011](#); [Sandbach & Lane-Serff 2011](#), and references therein) or distributed sources along a vertical wall such as those modelled with a constant flux of saline water by [Cooper & Hunt \(2010\)](#). [Cooper & Hunt \(2010\)](#) investigated a wall plume at a vertical boundary generated by injecting a saline solution through a porous sidewall of a square tank, thus providing a constant flux of buoyancy over the entire depth. The evolution of the stratification and the motion of the density front between the heated fluid and the homogeneous ambient in the box showed an overall good agreement of the experimental data with the entrainment model of [Morton *et al.* \(1956\)](#) that was adapted to a vertically distributed source along the wall.

For an isothermal wall the heat flux varies with height, since the rising fluid will be subject to a gradually smaller temperature difference with the sidewall. For the laminar case [Worster & Leitch \(1985\)](#) showed that there is no density front as was observed by [Baines & Turner \(1969\)](#) for a turbulent plume between the homogeneous environment and stratified fluid. Since the laminar boundary layer remains unmixed, it enhances a different stratification than a turbulent wall plume that engulfs and mixes ambient fluid of lower temperatures.

In the present study, we consider a heated wall that gives rise to a wall plume that is laminar near the base and turbulent above a certain height. A hybrid entrainment model, based on the model by [Worster & Leitch \(1985\)](#) for the laminar part, and on [Cooper & Hunt \(2010\)](#) for the turbulent part, is presented, where we consider a constant temperature wall instead of a constant buoyancy flux. With this modelling, we implicitly also respond to questions about the differences in Prandtl number, which is 5.5 for heat flux compared to a Schmidt number of 700 for salt flux experiments. Note that this modelling is one-dimensional as per the MTT model. Although two-dimensional models for laminar-turbulent wall plumes are available (see [Wells & Worster 2008](#)), we will continue the simpler one-dimensional approach.

The MTT model assumes there is a constant entrainment coefficient in the buoyant plume that is proportional to the mean local vertical velocity and that the buoyant plume fluid is ejected into the non-turbulent ambient (see e.g. [Wells *et al.* 1999](#), for an overview of the MTT model). The laminar model of [Worster & Leitch \(1985\)](#) is based on similarity solutions for a heated boundary layer. The transport of a plume generally causes a stratified top layer that increases in depth with time so that the ambient is partially stratified and partially homogeneous. Since the entrainment coefficient in shear layers is higher in a homogeneous fluid than in a stratified fluid (see [Fernando 1991](#)), one may expect that for a plume with constant buoyancy and velocity, the increase of the stratified layer-depth also causes the entrainment coefficient to decrease with time. Usually, a constant entrainment value that does not take into account this variation is considered (see [Cooper & Hunt 2010](#)). This provides an additional aspect in the modelling. Therefore, measurement results for the plume and the ambient stratification are confronted with the

MTT and hybrid plume model, both for a constant and subsequently variable entrainment coefficient.

Although turbulent wall plumes in closed cavities have been investigated numerically, most experimental studies are concerned with local point measurements with probes (see e.g. Tsuji & Nagano 1988). Detailed quantitative information from experiments about the entire temperature and velocity field in the plume and the ambient stratification is not available at present. To test the modelling, a novel experimental measurement technique is used which allows the simultaneous measurement of temperature and velocity in an entire field and reveals both the details of the plume and the interior stratification. The instantaneous temperature fields are obtained using laser induced fluorescence (LIF) and by taking advantage of the variation in the emission spectrum of certain fluorescent dyes at particular temperatures (Walker 1987). This temperature-LIF method (hereinafter called T-LIF) was first tested by Nakajima *et al.* (1991) for a local measurement and by Sakakibara *et al.* (1993) for a planar field. As far as we know, this method has not been exploited for convective flows, where quantitative information is still limited to point measurements with thermistors or hot wires. For Rhodamine B that is used here, one can expect a fluorescent emission decay of 3.2%/K under favourable conditions, which allows for a precision of 0.2°C which is close to the precision of a thermistor measurement in a point. Velocity fields are obtained using Particle Image Velocimetry (PIV).

The experimental method is described in detail in section §2. The observations of the plume motion and the interior stratification are discussed in section §3. The entrainment model theory for an isothermal wall is presented with the corresponding numerical models in section §4, and the variation in the entrainment coefficient is discussed. The comparison between experimental results and theoretical models is made in §5. Results are further discussed in the conclusions in section §6.

2. Experimental measurements

2.1. The experimental setup

Experiments were conducted in a 551 mm-high tank of horizontal section $588 \times 300 \text{ mm}^2$, filled with demineralized water at room temperature (see figure 1). The front and back sides of the tank were made of 20 mm-thick transparent plastic (polymethyl methacrylate, or PMMA), whereas 5 mm-thick aluminium plates at the sidewalls conducted the heat from the water in the adjacent compartments to the test section. In the present experiments, one lateral sidewall (volume $300 \times 20 \times 551 \text{ mm}^3$) was filled with hot water which was kept at a constant temperature by means of a thermostatic bath with a maximal power of 2.2 kW and a precision of 0.1°C. This water was circulating at a rate of 0.43 L s^{-1} . For insulation the other side as well as the bottom were covered respectively with PMMA and PVC, respectively, whereas the top boundary was covered with extruded polystyrene foam (see figure 1(a)). Since the aluminium wall was in contact with two liquid baths at different temperatures, its own temperature was situated in between and changed gradually as the tank warmed up. Although this slowed down the total heating time of the test section, and therewith the different phases of the flow evolution, this change was too slow to modify the flow dynamics. Before starting the experiment it was verified that the fluid in the test section was at rest. At the start of the experiment, hot fluid from the thermostatic bath was injected into the (empty) side compartment. To compensate for the initial transient effects of heating the system, the fluid injected at the start of the experiment was preheated to a few degrees above the desired temperature (T_c).

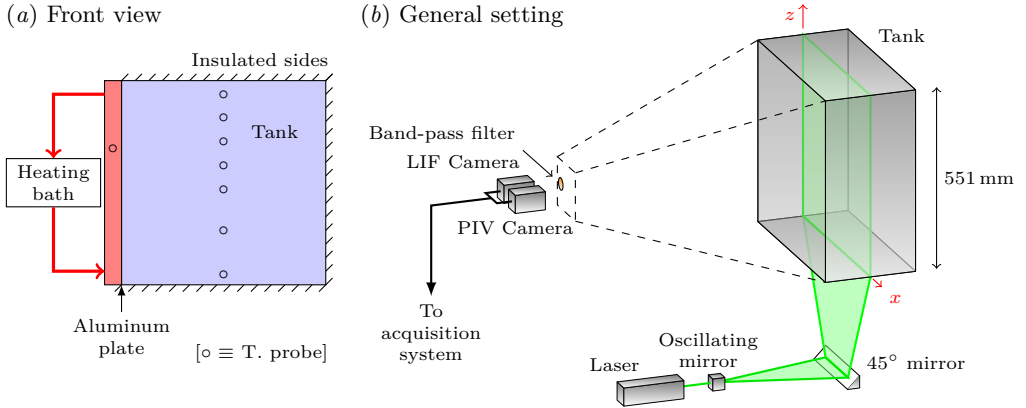


Figure 1: Sketch of the experimental setup with (a) the heating system and (b) the PIV/LIF image acquisition system.

Experiments were conducted at the initial temperatures of $T_0 = 21.4^\circ\text{C}$ in the test section and at $T_c = 53.6^\circ\text{C}$ in the lateral compartment, the latter being controlled by a probe. This was the maximum temperature difference supported by the experimental device, which was sufficient to get a significantly turbulent plume. The resulting wall temperature was approximately 39°C (see appendix A for details on the calculation). Therefore the Rayleigh number and the Prandtl number for our experiments are respectively,

$$\text{Ra}_H = \frac{g\beta(T_c - T_0)H^3}{\nu\kappa} \approx 5.4 \times 10^{10} \quad \text{and} \quad \text{Pr} = \frac{\nu}{\kappa} \approx 5.5, \quad (2.1)$$

where β , ν and κ are the fluid thermal expansion, kinematic viscosity and thermal diffusivity, respectively (taken at 30°C), g is the gravitational acceleration and H the height of the cavity. The compartment temperature was kept constant in the present paper, and a dozen tests were conducted for this same temperature difference, with very similar results obtained in all cases.

To monitor the temperature and calibrate the T-LIF images in the test section, the temperature was measured near the rear wall in the middle of the tank with 7 to 11 platinum temperature probes (see figure 1(a)) with a precision between 0.2 and 0.5°C . To visualize the particles and the fluorescent dye, the flow was illuminated by a vertical laser sheet shining through a transparent slit at the bottom. A solid continuous laser (Spectra-Physics) of 532 nm operating at 4.5 W was used and generated a beam of 2.3 mm thickness which was transformed into a light sheet by means of an oscillating mirror. Considering the relatively slow velocities in the convection plumes, an oscillating mirror is preferable to a cylindrical lens since it generally provides a more uniform sheet. To ensure identical lighting on every image and minimal exposition of the dye to the laser light, the mirror oscillations were triggered in phase with the image acquisition so that the laser beam passed an integer number of times over the entire view during each exposure time. For temperature measurements with LIF, Rhodamine B was used as fluorescent dye. Details are discussed below (§2.2). For PIV measurements, the water was seeded with polyamide tracer particles with a mean diameter of 30 μm and a density of 1016 kg m^{-3} .

The flow was recorded simultaneously by two 12-bit 1024×1024 pixel Dalsa CCD cameras at a frame rate of 1 Hz. One camera only recorded the particle positions for PIV

measurements by taking bursts of three successive images, and the other camera was equipped with a band filter with wavelengths passing between 565 to 585 nm and recorded the fluorescent dye for the T-LIF measurements. To compute the velocity fields for each burst, two pairs of images representing respectively a short and a long time interval, were processed using the PIV cross-correlation algorithm Uvmat (Matlab toolbox <http://servforge.legi.grenoble-inp.fr/projects/soft-uvmat>). Thus, we obtained a field containing 17 vectors per square centimetre with an average error estimated at 6% in the plume region.

All the variables employed in this paper will be described within a Cartesian coordinate system (x, z) with z being the upward vertical direction (see figure 1(b)).

2.2. Details of the T-LIF technique

Rhodamine B is a temperature sensitive dye in which the emission sensitivity to temperature changes with wavelength. Dissolved in water, the emission peak is around 575 nm, and the sensitivity to temperature is at its maximum around 570 nm (see Bruchhausen *et al.* 2005). An optimal choice for the bandpass filter is found at 565 to 585 nm. To minimize absorption of incident light along the laser path, the Rhodamine B concentration was chosen to be as low as possible. Since the dimensions of the experiment are significantly bigger than the dimensions used in previous T-LIF applications (Coolen *et al.* 1999; Bruchhausen *et al.* 2005; Petracci *et al.* 2006), we used a relatively lower concentration here, i.e. $3.5 \times 10^{-5} \text{gL}^{-1}$.

In order to avoid permanent local intensity variations due to light absorption or optical aberrations (see figure 2(a)), the images were normalized with two reference fields taken for two homogeneous temperatures T_{lo} and T_{hi} (see Coolen *et al.* 1999):

$$I_{\text{n}}(x, z) = \frac{I(x, z) - I_{\text{hi}}(x, z)}{I_{\text{lo}}(x, z) - I_{\text{hi}}(x, z)}. \quad (2.2)$$

This equation allows us to find a unique polynomial relation between light intensity and temperature. To obtain this relation, each experiment has been calibrated in advance, using the temperature data from the probes in the tank. An example of this procedure is given in figure 2(b). The final result (see figures 3(a) and (b)) has a precision of approximately $\pm 0.2^\circ\text{C}$ (figure 3(c)) on a field of 3200 cm^2 corresponding to 1024×1024 pixels (see figure 3), which is relatively high compared with previous T-LIF experiments reported in the literature.

Note that other methods referred to as ‘two-colour’, based on two dyes (e.g. Coppeta & Rogers 1998) or two spectral bands of the same dye (e.g. Bruchhausen *et al.* 2005), may be used to correct for variations in the illumination. However the dimensions of our experiment make them very difficult to apply here because of light absorption and re-absorption phenomenon, and, given achieved accuracy of the simpler one-colour method, the expected gain should not be significant.

3. Observations from T-LIF & PIV measurements

3.1. Plume and stratification evolution

At the start of the experiment the heated wall causes a thin hot plume which rises along the boundary and hits the top of the tank (figures 4(a)). A vortex motion forms in the top left-hand corner and, while moving to the right along the top boundary, entrains ambient cold fluid at rest (figure 3(a) and 4(b)). When it hits the right wall, it moves vertically downwards against buoyancy, transferring its energy from inertia into mixing.

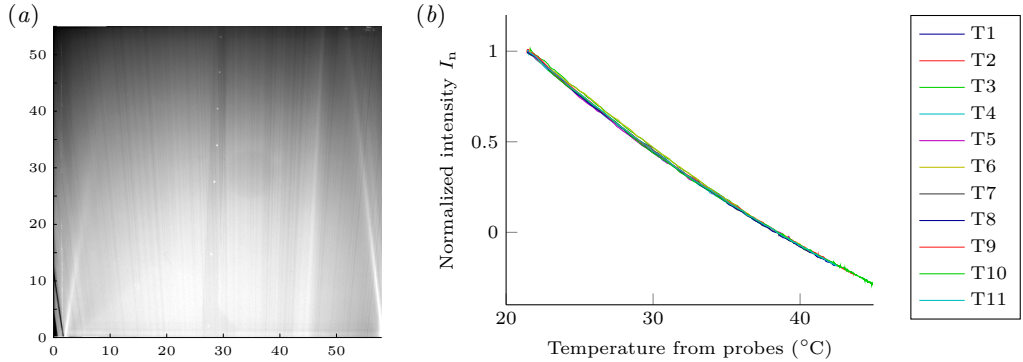


Figure 2: T-LIF calibration with (a) raw image (axes in cm) taken at constant temperature and showing fluorescence intensity variations due to absorption along beam paths and optical reflections. These defaults are corrected by normalization (see text). (b) Normalized intensity I_n at the position of the vertically distributed temperature probes plotted against their measured temperature. The 11 resulting curves are all fitted by the same 2nd order polynomial (in this case: $I_n = 6.91 \times 10^{-4}T^2 - 1.01 \times 10^{-1}T + 2.86$).

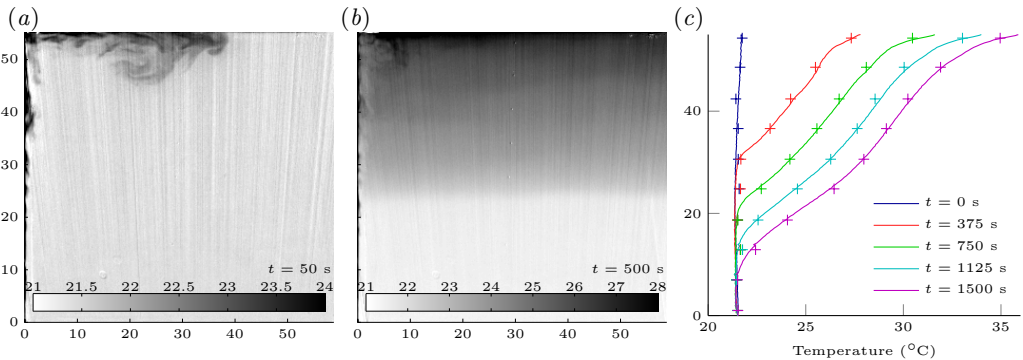


Figure 3: Examples of T-LIF results. Gray levels represent temperature in Celsius degrees, axes are in centimetres. (a) Temperature field at the beginning of the experiment ($t = 50$ s). (b) Temperature field when stratification is established ($t = 500$ s). Remaining striations due to temporal variations in laser light did not affect the results. (c) Stratification profiles taken at different times, averaged over 10 s with probe measurements (+) and T-LIF measurements (solid lines).

A gravity-current-like motion results and returns the mixed fluid until it hits the left wall, whereafter the newly formed warm upper layer makes a slumping wave-like motion (figures 4(c),(d),(e)). This overturning and slumping motion has also been observed for plumes at the centre of a cylindrical tank, with the flow evolution scenario depending on the tank aspect ratio (Kaye & Hunt 2007).

In the next stage of the flow evolution, the wall plume continues to supply hot fluid to this warm upper layer. This layer is now of a slightly lower temperature than that in the plume. In the next stage, instead of a vortex the plume motion gives rise to a jet-like motion along the top boundary. With the thickening of the upper layer the local buoyancy decreases and this jet also decreases in intensity as can be inferred from the vectors in figures 4(g)-(i). The continuous supply of hot fluid above formerly injected fluid causes

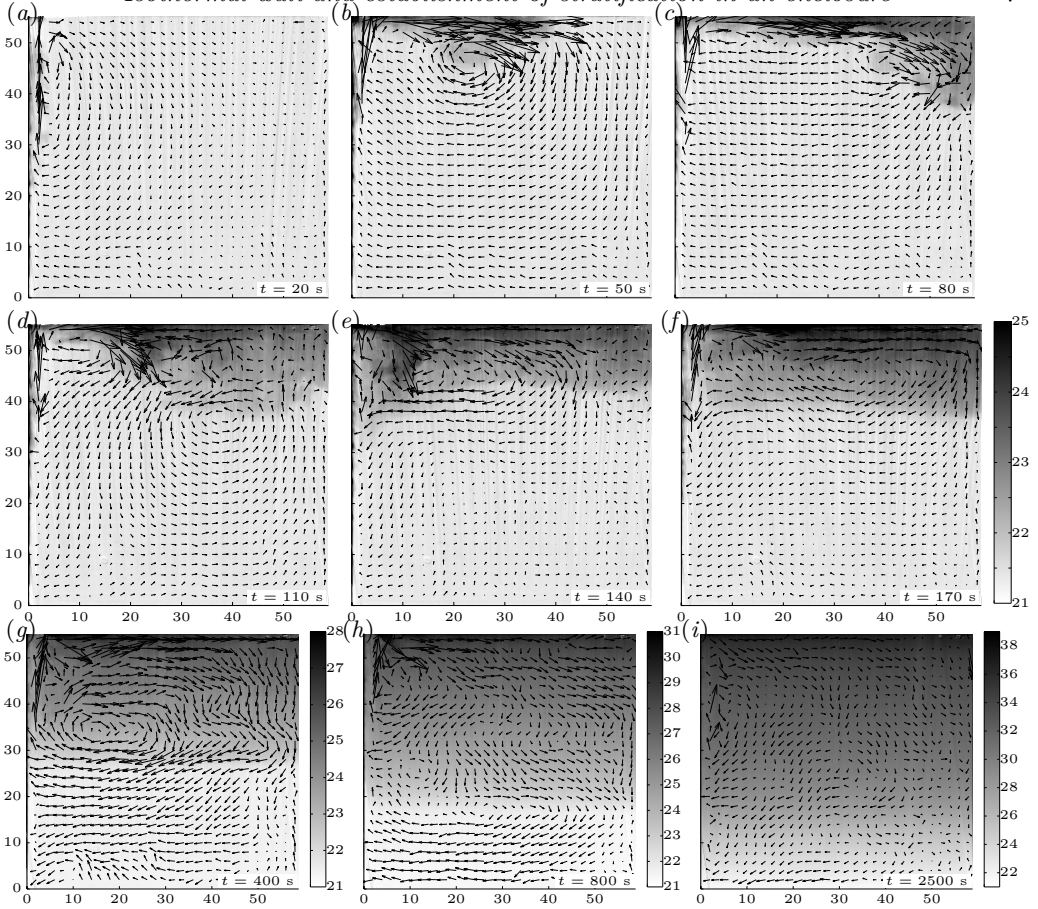


Figure 4: Sequence of images showing the onset of the turbulent plume and the establishment of the stratification. Gray levels indicate the temperature and arrows the velocity field. Axes are in centimetres, and temperature is in degrees Celsius. (a)-(f) show fields at the beginning of the experiment (velocity scale $1 \text{ cm} : 0.1 \text{ cm s}^{-1}$). (g)-(i) show fields at later times (velocity scale $1 \text{ cm} : 0.05 \text{ cm s}^{-1}$).

the descent of the interface. Eventually, after approximately 40 min of heating, the warm upper layer reaches the bottom, and the interior is entirely stratified in temperature (figure 4(g)-(i)).

In the subsequent final stage, the plume entrains fluid from the stratified ambient and gradually decreases the interior temperature gradient. The temperature profiles shown in figures 3(c) and 5 reveal the details of the evolution of the stratification and the gradual heating of the interior. These temperature profiles remain very similar in shape up to approximately 2000 s, i.e. before the warm layer has reached the lower boundary. The temperature profile near the bottom is subsequently distorted because of the laminar character of the plume in this region.

From the above evolution we can distinguish three dynamically different zones: (A), (B) and (C) (see figure 5). Zone (A) at the top ($z > 50 \text{ cm}$) is characterized by a strong stratification in a thin top layer that is continuously supplied by hot and hardly mixed plume fluid. This warm top layer (A) forms at the very beginning and was also present in the experiments with salt plumes of Cooper & Hunt (2010). Closer inspection of the plume

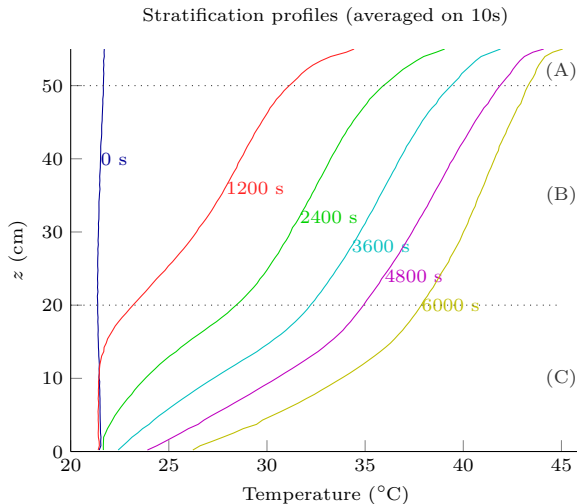


Figure 5: Ambient temperature profiles at different times showing the evolution of stratification.

motion below shows that this is due to the thin laminar flow adjacent to the vertical wall. This flow has been modelled in detail by [Wells & Worster \(2008\)](#). The middle zone (B) ($20 < z < 50$ cm) is stratified, and exchanges fluid with the wall plume. In the lower layer, zone (C), the plume is thin and laminar. As a consequence, the exchange of fluid and the mixing of the layers is weak. Since the heated layers are pushed downwards by continuity, the temperature gradient in this region increases in time. Eventually, zones (A), (B) and asymptotically also (C) will become homogeneous and obtain the temperature of the heated sidewall.

To study the turbulent properties of the plume, we consider the Reynolds decomposition $u_i = \bar{u}_i + u'_i$, with the bar denoting a time average over 200 s and the prime representing fluctuations. The Reynolds tensor is defined as $\overline{u'_x u'_z}$ and is displayed in figures [6\(a\)-\(c\)](#) at three different instants in time (i.e. respectively at $t = 80, 400$ and 1600 s), with the highest values at the edge of the plume, indicating turbulent mixing. A clear transition is visible at a height in the range of 22 to 25 cm. This transition corresponds to a change in flow regime from laminar to turbulent, as can be evaluated using the local Grashof number defined as

$$\text{Gr}_z = \frac{g\beta\Delta T z^3}{\nu^2}, \quad (3.1)$$

where ΔT is the temperature difference between the temperature imposed at the wall, and the temperature outside the plume. For $\Delta T \approx 15^\circ\text{C}$, the typical transition value of 10^9 is actually reached at $z_{\text{crit}} \approx 24$ cm ([Bejan & Lage 1990](#)). Indeed a clear turbulent activity can be noticed in figures [6\(a\)-\(c\)](#) above this height. This transition from laminar to turbulent flow in the boundary layer is noticeable in all measurements.

The turbulence and consequently the mixing starts in zone (B). During the flow evolution it also continues to mix when the warm layer has increased in thickness (see figures [6\(a\)-\(c\)](#) in relation to [6\(d\)-\(f\)](#)). In contrast, close to the wall the plume is laminar and heated fluid is transported upward without much mixing and causes the high temperature gradient near the top in zone (A). The large temperature gradient perpendicular to the wall is clearly noticeable in figures [6\(d\)-\(f\)](#).

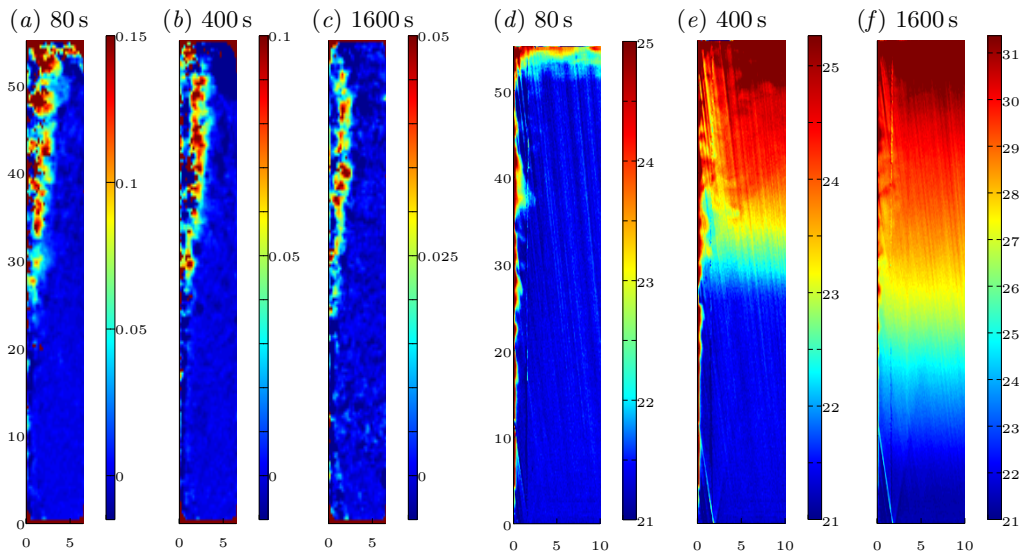


Figure 6: (a)-(c) Reynolds tensor $\overline{u'_x u'_z}$ ($\text{cm}^2 \text{s}^{-2}$) and (d)-(f) temperature ($^\circ\text{C}$) in the plume region at respectively $t = 80, 400$ and 1600 s after the start of the forcing. Axes are in centimetres.

3.2. Ambient flow characteristics

More detailed information about the flow can be obtained from the local Richardson number, defined as

$$\text{Ri} = -\frac{g}{\rho(x, z)} \frac{\partial \rho(x, z)}{\partial z} \bigg/ \left(\frac{\partial U(x, z)}{\partial z} \right)^2, \quad (3.2)$$

where $\rho(x, z)$ and $U(x, z)$ are the local density and velocity, respectively. In general, the Richardson number reveals gravitational instabilities ($\partial \rho(x, z)/\partial z > 0$) and shear or Kelvin-Helmholtz instabilities ($0 < \text{Ri} < \frac{1}{4}$). Gravitational instabilities are very localized to instantaneous and intermittent flow reversals during the initial flow stage (see figure 4(b)). Therefore, here we focus on shear instabilities that are displayed for visualization purposes on a logarithmic scale of the absolute Richardson number in figure 7). The details of the mixing are highlighted by the low (dark) values of the Richardson number. In particular the mixing due to the continued plume motion along the top boundary is important during the upper layer formation and its growth in time (see figures 7(a)-(f)). After the upper layer formation in the initial flow stage, the mixing occurs principally between the rising plume and the interior, whereas the mixing between the upper and lower layer is negligible (see figure 7(e)-(g)).

The streamlines and stratification in figure 8 show the averaged flow evolution of the circulation during the warm layer deepening. As in the case of a laminar plume (see Worster & Leitch 1985, figure 7), the hot fluid injection along the top boundary causes the entire stratification to move downwards. The difference with laminar flow is apparent from the turbulent exchange of the warm layer with the plume motion (top-left quarter), and becomes more pronounced with time from the circulation pattern in figures 8(b) and (c). This exchange and consecutive mixing in the plume is responsible for the presence of the front between the upper and lower layer. This front is indeed absent in the case of a laminar plume. In the lower layer, by continuity, the descending interface forces the

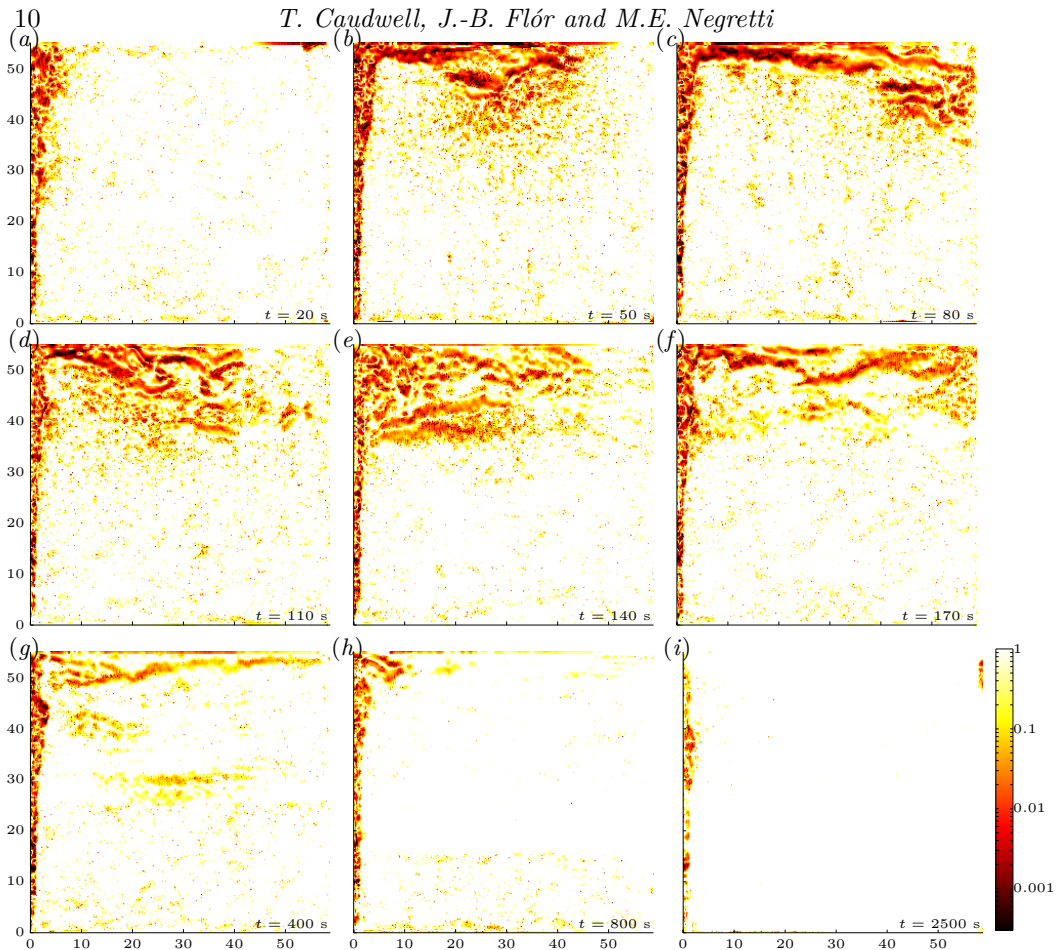


Figure 7: Absolute value of Richardson number $Ri = \left| -\frac{g}{\rho} \frac{\partial \rho}{\partial z} / \left(\frac{\partial U}{\partial z} \right)^2 \right|$ at different times corresponding to figure 4 using a logarithmic scale ($Ri > 1$ in white zones). Black spots in the unstratified zone (C) ($\frac{g}{\rho} \frac{\partial \rho}{\partial z} \approx 0$) are due to noise. Axes are in centimetres.

fluid in the lower layer into the plume. This effect becomes more pronounced once the lower layer is thinner and the motion remains horizontal (figure 8(c)).

3.3. Wall plume details for the modelling

Figure 9(a) shows the velocity and temperature profiles in the plume. To reduce the scatter, the data (recorded with a frequency of 1 Hz) are averaged over a period of 30 s. These velocity and temperature profiles conserve their general shape at all heights. To determine the characteristic quantities of the plume which are used in the model, i.e. the plume width $b(z)$, the characteristic plume velocity $w(z)$ and the characteristic plume temperature $T(z)$, we have used conservation of volume, momentum and energy (buoyancy) fluxes. They can be written for all z as

$$b(z)w(z) = \int_0^\infty u_z(x, z) dx, \quad b(z)w(z)^2 = \int_0^\infty u_z(x, z)^2 dx \quad (3.3)$$

$$\text{and } b(z)w(z)(T(z) - T_e(z)) = \int_0^\infty u_z(x, z)(T_{xz}(x, z) - T_e(z)) dx, \quad (3.4)$$

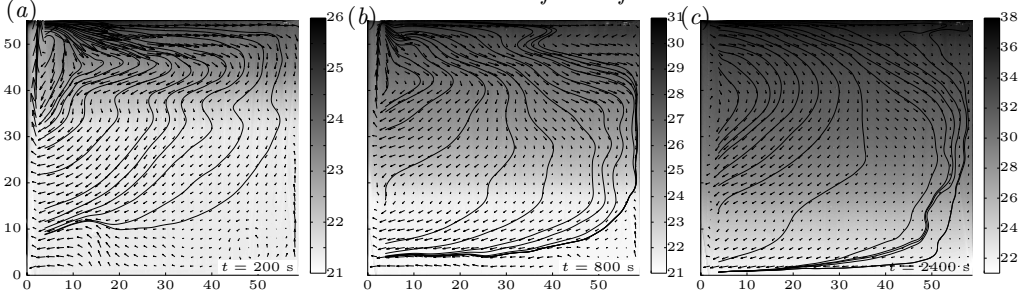


Figure 8: Velocity (arrows) and temperature (gray levels) fields averaged over 200 s, with streamlines patterns. Axes are in centimetres, temperature is in Celsius degrees. Time averages cover the spans (a) $t = 100 - 300$ s, (b) $t = 700 - 900$ s and (c) $t = 2300 - 2500$ s.

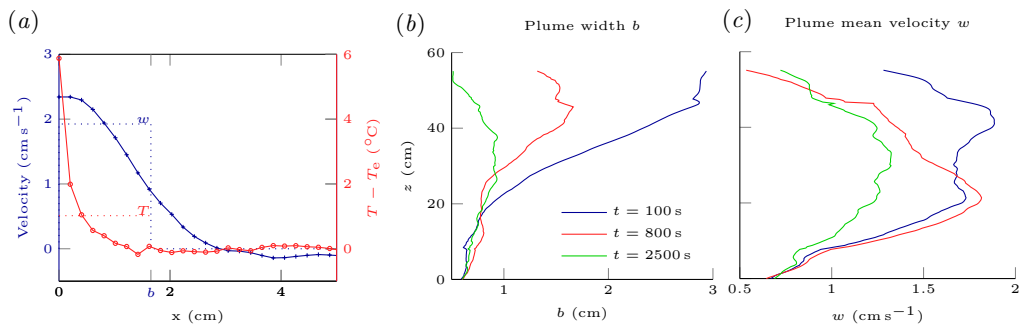


Figure 9: Averaged quantities in the plume. (a) Typical vertical velocity and temperature profiles (at $z \approx 40$ cm and $t = 150$ s). The dotted lines show the top-hat profile used in the model (§4). (b) Plume width b as a function of z at different times. (c) Plume characteristic velocity w as a function of z at the same times.

where $u_z(x, z)$ and $T_{xz}(x, z)$ are respectively the local vertical velocity and temperature, respectively, and $T_e(z)$ is the ambient temperature. To simplify notation, $b(z)$, $w(z)$ and $T(z)$ will be written as b , w and T , respectively, in the following.

The profiles of b and w (in figures 9(b) and (c), respectively) clearly show the laminar-turbulent transition in the wall plume at $z \approx 24$ cm in accordance with figures 6(a)-(c). For $2 < z \lesssim 24$ cm the plume width remains almost constant while velocity increases linearly, and it expands in the zone for $24 < z < 45$ cm (see also figures 6(a)-(c)). The increase in plume width is not balanced by an equivalent gain in volume flux, so that the velocity w decreases. At the upper boundary the vertical volume flux goes to zero, and the plume is deviated horizontally. Since the vertical velocity also changes to zero, the evaluation of b becomes meaningless, as is apparent at $z > 45$ cm (figure 9(b)).

3.4. The entrainment coefficient

The entrainment coefficient is the key parameter for the modelling described below and will be determined empirically. It is evaluated with the equation of volume conservation (4.1) introduced in next section:

$$\alpha = \frac{1}{w} \frac{d(bw)}{dz}. \quad (3.5)$$

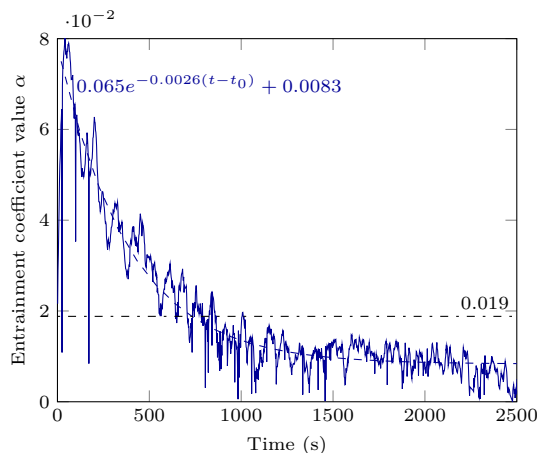
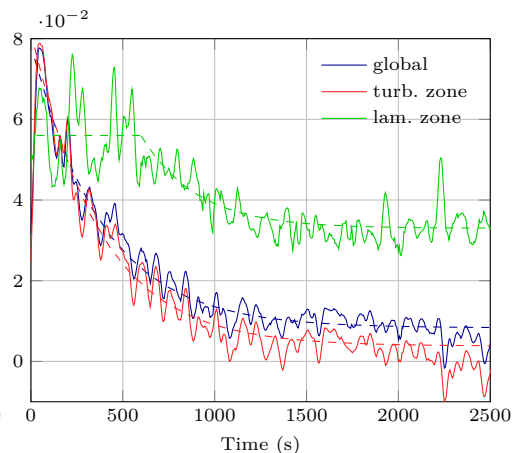
(a) Global evaluation of α (b) Evaluation of α per zone

Figure 10: (a) Experimental entrainment coefficient α calculated with (3.5) for the entire plume against time, with the average value (dash-dotted line) and exponential fit (dashed line) with t_0 taken at 30s for the onset time of the plume. (b) Entrainment coefficient, α , for the turbulent part of the plume (red), the laminar part (green) and total plume (blue) with the respective exponential fits. The data has been averaged over 20 s.

The experimental values for bw and w are accurately obtained from PIV measurements. Using a linear regression between $\frac{d(bw)}{dz}$ and w , a unique value for α is calculated at each time step for the entire wall plume and therefore includes the contributions of both the homogeneous and the stratified zones of the turbulent plume (see appendix B for details). The values are plotted in figure 10(a) against time. It starts around 0.08 when the environment is still homogeneous, a value which is close to that of a buoyant jet (e.g. Ellison & Turner 1959; Kaminski *et al.* 2005) or a fountain (e.g. Baines 2002). The time-averaged value $\alpha = 0.019$ is in good agreement with the entrainment value obtained by Cooper & Hunt (2010), which was obtained from the position of the density front. Although their value is close to the present one for $t > 1000$ s, there is a non-negligible decrease of a factor close to 10 during the establishment of the stratification over the depth of the tank.

The consequent raise in temperature of the interior also decreases the buoyancy flux of the plume. The mean Reynolds stress term therewith decreases in time by approximately a factor four, but the height of the transition between laminar and turbulent regimes characterized by the Grashof number hardly moves (see figure 6). This indicates that the turbulence activity of the plume decreases in time, but that it remains nevertheless turbulent. As mentioned above, in shear flows the entrainment coefficient in a stratified fluid is relatively small compared with that in a homogeneous fluid (see Fernando 1991). Since variations in time of buoyancy and velocity in the plume remain limited, it is the increase in stratified layer-depth in the ambient which decreases the entrainment coefficient.

Figure 10(b) differentiates the evolution of α in the turbulent (B) and laminar zones (A). (Note that the mechanism of entrainment in the laminar case is essentially different, but the definition (3.5) can still be used in this zone.) The initial rapid decrease of α in the turbulent zone is due to the increasing stratified layer-depth, with the minimal value

reached when eventually the interior is entirely stratified (see figure 7(i)). In the laminar zone, α remains constant at the beginning as the density of the ambient does not change, and then it decreases after approximately 600 s, when the front reaches the laminar part of the plume.

The sampling of the present data is not sufficient to calculate the local entrainment coefficient. However, figure 5 shows that the global temperature gradient in the turbulent zone (B) remains approximately constant (see figure 10(b) after 1200 s). The entrainment coefficient within this stratified zone is also found approximately constant. The two constant values for α , for the homogeneous and the stratified regions, allow the decrease of α observed in the turbulent zone to be accurately reproduced. The details of this calculation are shown in appendix B.

4. Theoretical modelling

In order to model the heated boundary, we consider a vertical distributed source of buoyancy and herein closely follow the approach of Cooper & Hunt (2010) based on the plume theory that is originally developed by Morton *et al.* (1956). The main differences in the present approach are that the temperature rather than heat flux is imposed and that the heated wall is placed in a closed cavity without openings.

4.1. The turbulent conservation equations

We consider an adiabatic prismatic box of height H , width L (along y) and horizontal cross-sectional area A . At one sidewall, a distributed buoyancy source emits a buoyancy flux per unit area ϕ . The plume is considered to be fully turbulent and, in accordance with Morton *et al.* (1956) theory, we assume the Boussinesq approximation, and the similarity of the velocity and buoyancy profiles with height. Furthermore, the velocity profile is assumed to be top-hat and the rate of entrainment proportional to the velocity at that height. All water properties except density are assumed to be constant with temperature.

The width of the plume, its vertical velocity and relative buoyancy are respectively denoted as b , w and Δ . Δ can be expressed as $\Delta = g(\rho_e - \rho)/\rho_1 = g\beta(T - T_e)$. Subscripts ‘e’ and ‘1’ refer to ambient and reference fluid, respectively, whereas variables with no subscript refer to the plume. The reference variables may conveniently correspond to the characteristics of the fluid at $t = 0$. Let us also introduce the ambient buoyancy $\Delta_e = g(\rho_e - \rho_1)/\rho_1 = g\beta(T_1 - T_e)$. The conservation equations for volume, momentum and buoyancy deficiency in the plume are written as

$$\frac{d(bw)}{dz} = \alpha w, \quad \frac{d(bw^2)}{dz} = b\Delta - \varepsilon \quad \text{and} \quad \frac{d(bw\Delta)}{dz} = bw\frac{\partial\Delta_e}{\partial z} + \phi \quad (4.1)$$

where ε accounts for dissipation due to wall shear stress that we will neglect in the following.

Unlike in Cooper & Hunt (2010), the buoyancy flux, ϕ , is variable due to the heat flux, φ , across the wall. Their relation per unit area across the wall is given by

$$\phi = \frac{g\beta}{\rho_1 C_p} \varphi \quad (4.2)$$

where C_p is the specific heat capacity of water at constant pressure. The heat flux, φ , has to be defined here as a function of the temperature imposed at the wall T_w and the temperature of the ambient fluid T_e . Then we write

$$\varphi = h_w(T_w - T_e), \quad (4.3)$$

where h_w is the heat transfer coefficient which can be expressed according to Tsuji & Nagano (1988) as

$$h_w = K_w k \left(\frac{g\beta(T_w - T_e)}{\nu\kappa} \right)^{\frac{1}{3}}, \quad (4.4)$$

where the constant $K_w = 0.120$ and k is the thermal conductivity of the fluid. Since the buoyancy term for the ambient was written as $\Delta_e = g\beta(T_1 - T_e)$ where T_1 is the reference temperature, one can introduce a wall buoyancy, Δ_w , defined as $\Delta_w = g\beta(T_w - T_1)$, so that (4.2) becomes

$$\phi = A_w^{\frac{1}{3}} (\Delta_e + \Delta_w)^{\frac{4}{3}}, \quad (4.5)$$

where $A_w = K_w^3 \frac{\kappa^2}{\nu}$ is the effective diffusion constant quantifying the buoyancy flux from the wall. Re-writing the conservation equations in terms of volume flux $Q = bw$, momentum flux $M = bw^2$ and buoyancy flux $F = bw\Delta$, the equations (4.1) reduce to:

$$\frac{dQ}{dz} = \alpha \frac{M}{Q}, \quad \frac{dM}{dz} = \frac{QF}{M} \quad \text{and} \quad \frac{dF}{dz} = Q \frac{\partial \Delta_e}{\partial z} + A_w^{\frac{1}{3}} (\Delta_e + \Delta_w)^{\frac{4}{3}}. \quad (4.6)$$

In ‘‘filling-box’’ models the characteristic time scales of the ambient are assumed to be much larger than those in the plume. The heat equation in the ambient fluid is then written as

$$\frac{\partial \Delta_e}{\partial t} = -U \frac{\partial \Delta_e}{\partial z} + \kappa \frac{\partial^2 \Delta_e}{\partial z^2}, \quad (4.7)$$

where U is the vertical velocity of the ambient. Following Baines & Turner (1969)’s assumption for which the plume width is negligible compared with the horizontal cross-section of the box (i.e. $b \ll A/L$), U can be expressed as $U = -QL/A$.

We non-dimensionalize equations (4.6) and (4.7) with the scalings

$$\begin{aligned} \zeta &= H^{-1}z, & \tau &= LA^{-1}H^{-1}A_w t, & \delta_e &= H^3 A_w^{-2} \Delta_e, \\ f &= H^3 A_w^{-3} F, & q &= A_w^{-4} Q, & m &= H A_w^{-2} M \end{aligned} \quad (4.8)$$

and obtain for the conservation equations

$$\begin{aligned} \frac{dq}{d\zeta} &= \alpha \frac{m}{q}, \\ \frac{dm}{d\zeta} &= \frac{qf}{m}, \\ \frac{df}{d\zeta} &= q \frac{\partial \delta_e}{\partial \zeta} + (\delta_e + \delta_w)^{\frac{4}{3}}. \end{aligned} \quad (4.9)$$

Here the wall buoyancy term, δ_w is

$$\delta_w = K_w^{-6} \text{Pr}^3 \text{Ra}_H, \quad (4.10)$$

and the Rayleigh and Prandtl numbers introduced are defined as

$$\text{Ra}_H = \frac{g\beta(T_w - T_1)H^3}{\nu\kappa} \quad \text{and} \quad \text{Pr} = \frac{\nu}{\kappa}. \quad (4.11)$$

For the heat equation (4.7) we obtain in non-dimensional form

$$\frac{\partial \delta_e}{\partial \tau} = q \frac{\partial \delta_e}{\partial \zeta} + \frac{\text{Pr}}{K_w^3} \frac{A}{HL} \frac{\partial^2 \delta_e}{\partial \zeta^2} \quad (4.12)$$

where HL/A is the box aspect ratio. The second term on the right-hand side expressing molecular diffusion in the ambient is neglected in the following.

With equations (4.9) and (4.12) we can now determine the position of the first disturbed zone in the ambient, corresponding to the lower limit of the interface between the upper and lower layer (see [Worster & Huppert 1983](#)). For an initially uniform ambient, i.e. $\partial\delta_e/\partial\zeta = 0$, for the vertical position of the front, denoted ζ_{fr} , one obtains

$$\frac{d\zeta_{\text{fr}}}{d\tau} = -\frac{3}{4} \left(\frac{4}{5}\right)^{\frac{1}{3}} \alpha^{\frac{2}{3}} (\delta_e^{\text{fr}} + \delta_w)^{\frac{4}{9}} \zeta_{\text{fr}}^{\frac{4}{3}}, \quad (4.13)$$

where δ_e^{fr} is the initial ambient buoyancy. This can be integrated using the condition $\zeta_{\text{fr}}(\tau = 0) = 1$ to give

$$\zeta_{\text{fr}} = \left[1 + \frac{1}{4} \left(\frac{4}{5}\right)^{\frac{1}{3}} \alpha (\delta_e^{\text{fr}} + \delta_w)^{\frac{4}{9}} \tau \right]^{-3}. \quad (4.14)$$

Note that the front position in time depends directly on the entrainment coefficient α .

4.2. Hybrid model including the laminar part

As we have seen in the experimental results, the wall plume is laminar over approximately one-third of its height. This requires a particular treatment, especially for configurations with a lower Grashof number than the present one. Below, the variables with a hat are used for the modelling of the laminar zone. In order to measure its influence, similarity solutions are used for the stream function ψ and the temperature perturbation $T - T_e$, given a stratification profile proportional to some power of z (see [Worster & Leitch 1985](#)). Defining ψ such that $u_x = -\partial\psi/\partial z$ and $u_z = \partial\psi/\partial x$, these similarity solutions are written in terms of non-dimensional functions $\widehat{F}(\widehat{\xi})$ and $\widehat{G}(\widehat{\xi})$ as

$$\psi = \kappa(\lambda_T \text{Ra}_z)^{\frac{1-\hat{m}}{4}} \widehat{F}(\widehat{\xi}), \quad \text{and} \quad T - T_e = \lambda_T (T_w - T_1) (\lambda_T \text{Ra}_z)^{-\hat{m}} \widehat{G}(\widehat{\xi}). \quad (4.15)$$

The constants λ_T and \hat{m} characterize the density profile and satisfy the relation

$$\frac{T_w - T_e}{T_w - T_1} = \lambda_T^{1-\hat{m}} \text{Ra}_z^{-\hat{m}}. \quad (4.16)$$

(For the Rayleigh number [Worster & Leitch \(1985\)](#) use a characteristic temperature scale denoted ΔT , which is obtained from the instant density profile. To keep the Rayleigh number fixed, here ΔT is replaced by $\lambda_T (T_w - T_1)$, where the density variation is represented by the coefficient λ_T .) The ambient is homogeneous when $\hat{m} = 0$, and stably stratified for $\hat{m} > 0$. The two functions \widehat{F} and \widehat{G} are based on the similarity variable

$$\widehat{\xi} = (\lambda_T \text{Ra}_z)^{\frac{1-\hat{m}}{4}} \frac{x}{z}. \quad (4.17)$$

They satisfy the following two differential equations derived from the boundary-layer equations for steady convection along a plate:

$$\widehat{F}''' + \frac{1}{\text{Pr}} \left[\frac{3}{4} (1 - \hat{m}) \widehat{F} \widehat{F}'' - \frac{1}{2} (1 - 3\hat{m}) \widehat{F}'^2 \right] + \widehat{G} = 0 \quad (4.18)$$

$$\widehat{G}'' + \frac{3}{4} (1 - \hat{m}) \widehat{F} \widehat{G}' - 3\hat{m} (1 - \widehat{G}) \widehat{F}' = 0, \quad (4.19)$$

which are subject to the boundary equations

$$\widehat{F} = \widehat{F}' = 0, \quad \widehat{G} = 1 \quad \text{for } \widehat{\xi} = 0, \quad (4.20)$$

$$\widehat{F}' \rightarrow 0, \quad \widehat{G} \rightarrow 0 \quad \text{for } \widehat{\xi} \rightarrow \infty. \quad (4.21)$$

Using the non-dimensional variables (4.8), one can deduce the integral variables q , m and f for the laminar part ($\zeta < \zeta_{\text{crit}}$):

$$\begin{cases} q = K_w^{-3} \text{Pr}(\lambda_T \text{Ra}_H)^{\frac{1}{4}(1-\hat{m})} \widehat{K}_Q \cdot \zeta^{\frac{3}{4}(1-\hat{m})} \\ m = K_w^{-6} \text{Pr}^2(\lambda_T \text{Ra}_H)^{\frac{3}{4}(1-\hat{m})} \widehat{K}_M \cdot \zeta^{\frac{1}{4}(5-9\hat{m})} \\ f = K_w^{-9} \text{Pr}^4(\lambda_T \text{Ra}_H)^{\frac{5}{4}(1-\hat{m})} \widehat{K}_F \cdot \zeta^{\frac{3}{4}(1-5\hat{m})} \end{cases} \quad (4.22)$$

where \widehat{K}_Q , \widehat{K}_M and \widehat{K}_F are integrals calculated from the non-dimensionalized functions \widehat{F} and \widehat{G} as, respectively,

$$\widehat{K}_Q = \widehat{F}(\infty), \quad \widehat{K}_M = \int_0^{+\infty} \widehat{F}'(\widehat{\xi})^2 d\widehat{\xi} \quad \text{and} \quad \widehat{K}_F = \int_0^{+\infty} \widehat{F}'(\widehat{\xi}) \widehat{G}(\widehat{\xi}) d\widehat{\xi}. \quad (4.23)$$

These values rely on the constant \hat{m} which characterizes the stratification. Deducing the expression of the characteristic width and velocity, we obtain $b \propto \zeta^{(1-3\hat{m})/4}$ and $w \propto \zeta^{(1-3\hat{m})/2}$. Both agree well with observations shown respectively in figures 9(b) and (c). In the case of an ambient stratification composed of a lower homogeneous layer and an upper stratified layer, a similarity solution in the upper layer is still assumed unlike the influence of the lower layer. The set of equations (4.22) is then used two times, but with different values for \hat{m} and λ_T : the values for \hat{m} are zero in the homogeneous layer and positive above, and the values for λ_T are noted λ_T^h and λ_T^s , respectively. The similarity solutions obtained in the upper layer have to be corrected, since the fluxes from the unstratified lower layer are different from those that the stratified solution predicts. Thus the similarity variables $\widehat{\xi}$, \widehat{F} and \widehat{G} are adjusted in the upper layer by applying the coefficients λ_ξ , λ_F and λ_G , respectively, which are determined with the continuity conditions of the front. The continuity of the x -scale is written from (4.17) as

$$z_{\text{fr}}(\lambda_T^s \text{Ra}_{\text{fr}})^{\frac{\hat{m}-1}{4}} \lambda_\xi \widehat{\xi} = z_{\text{fr}}(\lambda_T^h \text{Ra}_{\text{fr}})^{-\frac{1}{4}} \widehat{\xi}, \quad (4.24)$$

giving

$$\lambda_\xi = C_\delta^{-\frac{1}{4}} \quad \text{with} \quad C_\delta = \frac{\lambda_T^h}{\lambda_T^s} (\lambda_T^s \text{Ra}_{\text{fr}})^{\hat{m}}. \quad (4.25)$$

Note that the constant C_δ embeds all the parameters necessary to describe the entire density profile. By a similar reasoning, the continuity of the characteristic velocity w , written as

$$\frac{\kappa}{z_{\text{fr}}} (\lambda_T^s \text{Ra}_{\text{fr}})^{\frac{1-\hat{m}}{2}} \frac{\lambda_F^2 \widehat{K}_M^s}{\lambda_\xi \lambda_F \widehat{K}_Q^s} = \frac{\kappa}{z_{\text{fr}}} (\lambda_T^h \text{Ra}_{\text{fr}})^{\frac{1}{2}} \frac{\widehat{K}_M^h}{\widehat{K}_Q^h}, \quad (4.26)$$

gives

$$\lambda_F = C_\delta^{\frac{1}{4}} \frac{\widehat{K}_M^h \widehat{K}_Q^s}{\widehat{K}_Q^h \widehat{K}_M^s}. \quad (4.27)$$

A third equality is derived from the continuity of the characteristic temperature inside the plume, which is proportional to the quantity $\delta - \delta_e$. Taking into account the density

jump in the ambient, equal to $\delta_e^{\text{fr}} - \delta_e^0$, we write

$$\frac{\kappa^2}{z_{\text{fr}}^3} \text{Pr}(\lambda_T^s \text{Ra}_{\text{fr}})^{1-\hat{m}} \frac{\lambda_F \lambda_G \widehat{K}_F^s}{\lambda_F \widehat{K}_Q^s} - \Delta_e^{\text{fr}} = \frac{\kappa^2}{z_{\text{fr}}^3} \text{Pr} \lambda_T^h \text{Ra}_{\text{fr}} \frac{\widehat{K}_F^h}{\widehat{K}_Q^h} - \Delta_e^0 \quad (4.28)$$

and obtain

$$\lambda_G = C_\delta \frac{\widehat{K}_Q^s}{\widehat{K}_F^s} \left[\frac{\delta_e^{\text{fr}} - \delta_e^0}{\lambda_T^h \delta_w} + \frac{\widehat{K}_F^h}{\widehat{K}_Q^h} \right]. \quad (4.29)$$

This development and the assumption of similarity for a partially stratified ambient have been successfully validated against experimental velocity fields (Caudwell 2015). This theory also allows the expression of the local value of α in the laminar zone:

$$\alpha(\zeta) = \frac{3}{4} (1 - \hat{m}) (\lambda_T \text{Ra}_H)^{\frac{\hat{m}-1}{4}} \frac{\widehat{K}_Q^2}{\widehat{K}_M} \zeta^{\frac{3}{4}(\hat{m}-1)}. \quad (4.30)$$

The overall value can then be computed from this expression and appears to compare well with the curve of figure 10(b) (not shown).

4.3. Temperature at the wall

In some cases the temperature T_w is known from measurements. Since it was not measured independently in the present experiments, we determine it from the temperature T_c in the side compartment and a global heat transfer coefficient h_g . This coefficient accounts for the transfer of heat across the aluminium wall from the thermostatic fluid to the working fluid. Because the heat flux is conserved throughout the vertical section, one may write

$$\varphi = h_w(T_w - T_e) = h_g(T_c - T_e), \quad (4.31)$$

providing an expression for the temperature at the wall as a function of T_c , T_e and the global heat transfer coefficient h_g . The resulting function can reasonably be reduced to a simple linear relation, given a limited range of variation for T_e (from 21 to 41°C). Details are given in appendix A. The averaged value of h_g has empirically been found to be equal to 305 W m⁻² K⁻¹. Since the thermal diffusivity of aluminium is 1000 times higher than that of water, we assume that δ_w is uniform within the plate, so that we obtain the relation

$$\delta_w = -0.567 \langle \delta_e \rangle + 0.544 \delta_c, \quad (4.32)$$

where $\langle \delta_e \rangle$ is the averaged value of the ambient buoyancy over the entire height. It is important to note that the variations of $\langle T_e \rangle$ are very slow compared with the dynamics of the flow, so that the variations of δ_w have no significance for the flow description.

4.4. Numerical procedure

Four models summarized in table 1 have been run to investigate the influence of both the laminar-turbulent modelling and the variation of α in time (source code available at <http://servforge.legi.grenoble-inp.fr/projects/soft-wpsm>). A first model, **Turb**, solves the equations (4.9) for a constant entrainment coefficient, using the average value of the plot represented in figure 10(a). A second model, **Hybrid**, takes into account the laminar part based on the similarity solution, and uses a constant α (mean value) for the turbulent zone. The third model, **Turb α** , uses a variable α following the exponential fit shown in figure 10(a). The fourth model, **Hybrid α** , includes both the laminar and the

Table 1: Summary of the different models presented with their respective parameters

Model	Entrainment coefficient α	Laminar model
Turb	0.019	No
Hybrid	0.014	Yes
Turbα	$0.065 \exp(-1.55 \times 10^7 \tau) + 0.0083$	No
Hybridα	$0.072 \exp(-2.05 \times 10^7 \tau) + 0.0038$	Yes

turbulent layer as in **Hybrid**, but has a variable entrainment coefficient for the turbulent layer based on the fit shown in figure 10(b).

To solve the system of equations (4.9) together with (4.12) we employ a numerical method similar to the one presented by Germeles (1975). Under the condition that the velocities within the plume are much larger than in the ambient, the numerical model solves the plume equations (4.9) at each time step. The slow movement of the stratification in the ambient follows (4.12), i.e. each layer moves downward by a small amount. In the model, the plume equations are discretized along ζ and solved by a Runge-Kutta method of fourth and fifth order, assuming that the δ_e and f profiles follow ‘staircase’ functions, i.e. they are constant on each step in space.

At each time step, δ_w is adjusted following (4.32). When the laminar zone is modelled, the value of ζ_{crit} is based on the instantaneous Grashof number. Once the stratification affects the laminar zone, the constants \hat{m} , \widehat{K}_Q , \widehat{K}_M and \widehat{K}_F are redefined at each iteration. Following (4.16), \hat{m} is determined by the plot of $\ln\left(1 + \frac{\delta_e}{\delta_w}\right)$ against $\ln(\text{Ra}_H \zeta^3)$ which has to be a straight line over the concerned zone for the similarity laws to be valid. It is verified that this is indeed the case.

The ambient is homogeneous at $t = 0$. For the fully turbulent model, the volume, momentum and buoyancy fluxes, q , m and f , respectively, are assumed to be equal to zero at the bottom boundary of the box. When the laminar zone is modelled, the equations (4.22) for $0 \leq \zeta \leq \zeta_{\text{crit}}$ are used to determine these quantities, and the calculation of the turbulent zone starts with these values at each iteration. Water properties such as thermal expansion β , kinematic viscosity ν , thermal diffusivity κ and thermal conductivity k are kept constant and evaluated at 30°C.

Figure 11 presents profiles for the ambient temperature, plume volume flux Q , and plume buoyancy flux F . Results are close to those of Cooper & Hunt (2010), except for the ambient temperature evolution which is reduced in our case as expected owing to the different boundary conditions. The volume and buoyancy fluxes are very similar, demonstrating that imposing a constant heat flux or a constant temperature is equivalent for a certain period of time, with the exception of the final temperature which is established within the cavity.

5. Comparisons between experiment and models

The four numerical models discussed in §4.4 have been run for the set of parameters presented in table 1 under conditions that are identical to the experiments. The experimental results (solid lines) are compared with the theoretical model (dashed lines) in figures 12 and 13. Before confronting the different models with the measurements, we recall that all models are one-dimensional. The horizontal redirection of the plume near the upper boundary is thus not incorporated. This has an influence on the density in

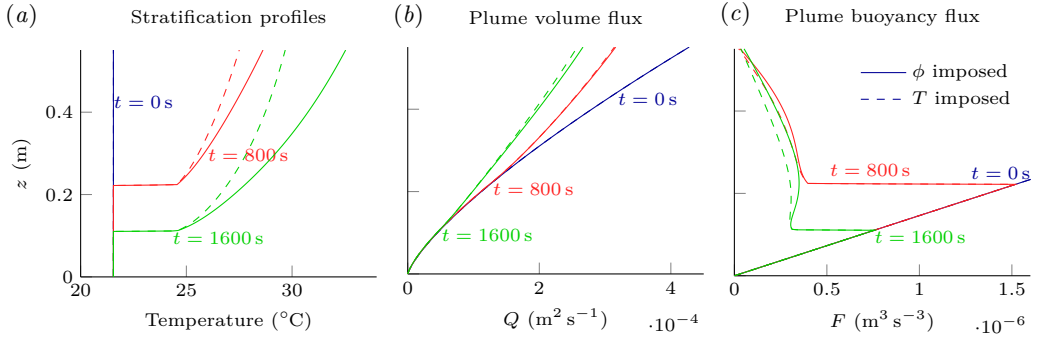


Figure 11: Comparison between the model with buoyancy flux ϕ imposed (solid lines) (Cooper & Hunt 2010) and constant wall temperature imposed (Turb, dashed lines). Variables are: (a) ambient temperature T_e , (b) plume volume flux Q and (c) plume buoyancy flux F . Both started with the same buoyancy flux $\phi = A_w^{\frac{1}{3}} \Delta_w^{\frac{4}{3}}$, with Δ_w defined in accordance with parameters of §3.4

the ambient that is not taken into account. The transition from laminar to turbulent regimes in the hybrid models is abrupt, and this causes a small discontinuity in the density profile that is more or less visible (see e.g. figures 12(d) and 12(h) at $z \approx 30$ cm and $t = 2000$ s). This appears when the stratification enters the laminar zone and causes subsequent changes in the laminar model ($\hat{m} \neq 0$).

Figures 12 shows the evolution of the temperature stratification in the ambient for each model. The vertical position of the isotherms is plotted against time in the left column, the lowest curve (22°C) representing the first front. The temperature profiles for different instants of time are represented on the right-hand side. The characteristic plume variables, i.e. buoyancy flux F , volume flux Q and momentum flux M , are displayed in the three columns of figure 13.

Note that in the experiment the origin of time has been delayed by approximately 30 seconds, so that $t = 0$ s corresponds to the time at which the turbulent plume is fully established, in coherence with the model assumptions. Below we discuss the evolution of the ambient stratification, as well as the characteristics of the plume.

5.1. The ambient stratification

All models predict the ambient stratification reasonably well. This confirms former experimental studies reported in the literature which successfully validate the MTT theory against the measured evolution of the stratification. Major discrepancies occur for the sharpness of the density front in models Turb and Hybrid for constant α . Small differences in the modelling also appear on all graphs near the top boundary due to the presence of a horizontal laminar layer in the experiment as shown in figures 12-right (explained at the end of §3.3).

As expected, the Hybrid model improves the agreement with the density profiles for small z . In particular figures 12(d) and 12(h) shows that the front attains the bottom of the cavity asymptotically, whereas it never does for the MTT modelling. This is an essential difference between the laminar and turbulent models (see Worster & Leitch 1985).

When including the variation of α with time, the difference between the Hybrid and the Turb models becomes more pronounced especially at the front. A high value for α

at $t = 0$ indeed reduces the density step. The variation of α also improves the overall agreement between the experimental and theoretical density profiles.

5.2. The plume

The simultaneous information about the density and the velocity in a field allows for the measurement of volume flux Q , momentum flux M and buoyancy flux F in the plume (see figure 13). These are not accessible from classical experiments. Generally, above $z = 42$ cm the deviation of the plume results in large differences with the models for Q and M (first and second columns), especially at early times. In particular the initial horizontal motion near the top after the onset of the plume and the consequent entrainment biases the comparison with the models (see figure 4(d)) so that the best comparison with theoretical models is obtained for $t > 100$ s. This artefact is absent for F (third column) since the buoyancy flux is set to zero at the top boundary. The introduction of a zero vertical velocity is not possible in the one-dimensional model because of the conservation of momentum. Furthermore, we notice that the volume and momentum fluxes, shown in the first and second columns of figure 13, respectively, have the same order of magnitude as found in the experiments.

The hybrid model, **Hybrid**, introduces an important change in the slopes of the momentum and volume fluxes in particular (see figures 13(e),(j) and (k) at $z \approx 24$ cm) corresponding to the laminar-turbulent transition. It shows much better agreement with the experiments for small z and improves the modelling of the turbulent region as well.

Although the variation of α alone (model **Turb α**) is of benefit to the modelling of Q (figure 13(g)), it does not significantly change M (figure 13(h)) and makes F even worse (figure 13(i)). **Hybrid α** models the volume flux Q (figure 13(j)) much better, and also shows a reasonable agreement for M (figure 13(k)). Therefore, the best results are obtained when both the laminar zone and the variable entrainment coefficient are included.

Strong discrepancies remain nevertheless for M and F in all cases. The former are most likely due to the shear stress along the wall, which is neglected in all cases. The poor agreement found for the buoyancy flux can be explained by the technical difficulty to evaluate it precisely in the experiment: because of the shape of the temperature profile within the boundary layer (see figure 9(a)), local lack of resolution or inaccurate positioning of the x -origin may result in significant underestimations of the buoyancy. In addition all fluid parameters except density are kept constant with the temperature, even though the mean temperature evolves in time. In particular F shows a strong dependency on the Prandtl number in the laminar regime (see equation (4.22)).

In conclusion, the introduction of the laminar zone mainly benefits the modelling of the plume and the lower part of the eventually established stratification. On the other hand the introduction of a variable α is important to properly model the overall density profile.

6. Conclusions

We have investigated a turbulent wall plume in a cavity induced by a hot isothermal wall and, using the novel T-LIF/PIV measurement method, we have obtained a detailed description of the wall-plume evolution, ambient stratification and internal circulation. We firstly summarize the most important experimental results. Observations reveal the relevance of a laminar zone near the base of the heated wall. The fact that the main mixing with the interior only takes place in the turbulent region, has an important influence on the shape of the stratification profile. The increasing stratification in the

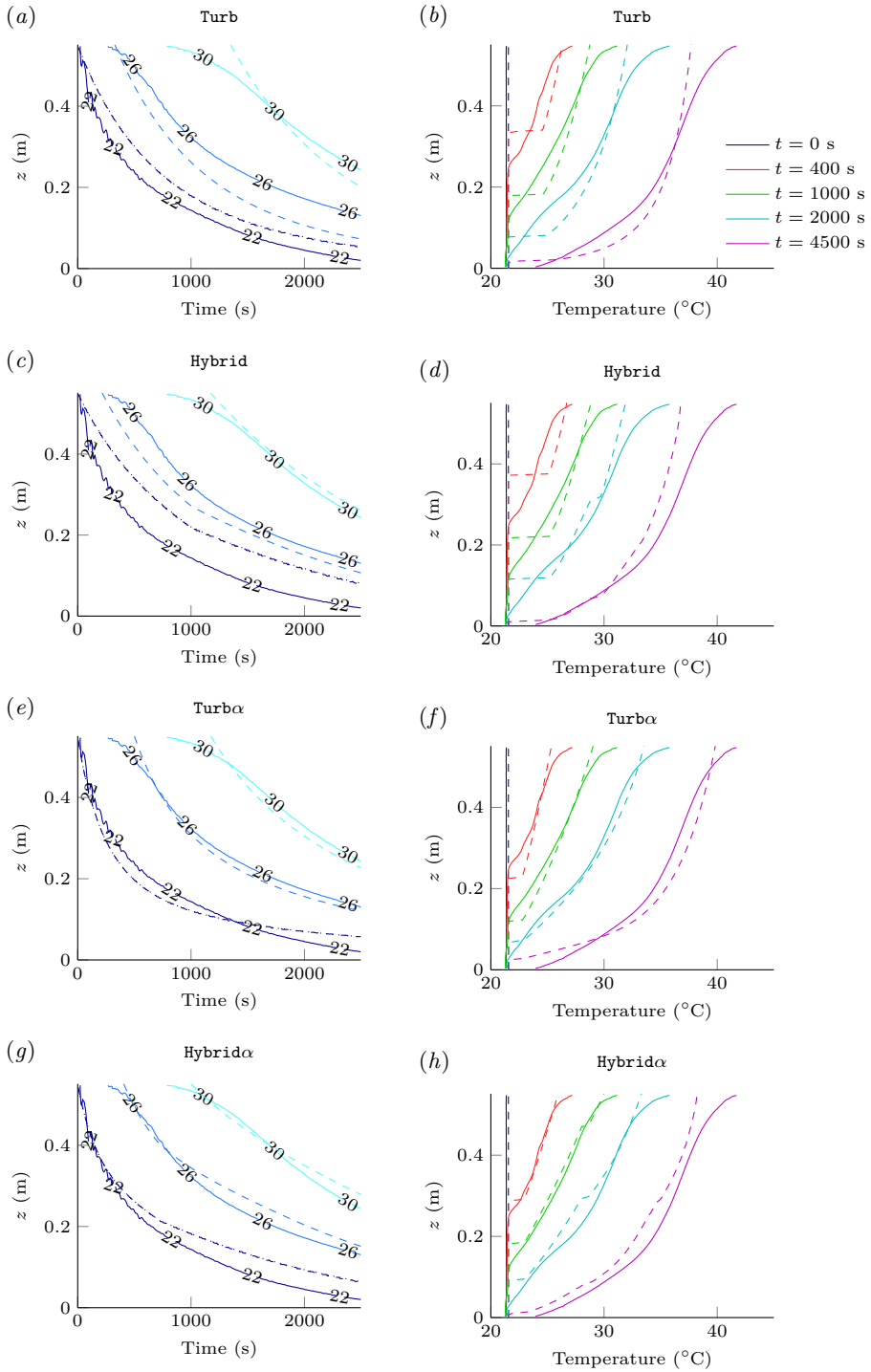


Figure 12: Evolution of the stratification in the ambient for experiments (continuous lines) and models Turb, Hybrid, Turb α and Hybrid α (dashed lines). On the left: the position of isotherms as a function of time. On the right: the stratification profiles at the centre of the tank.

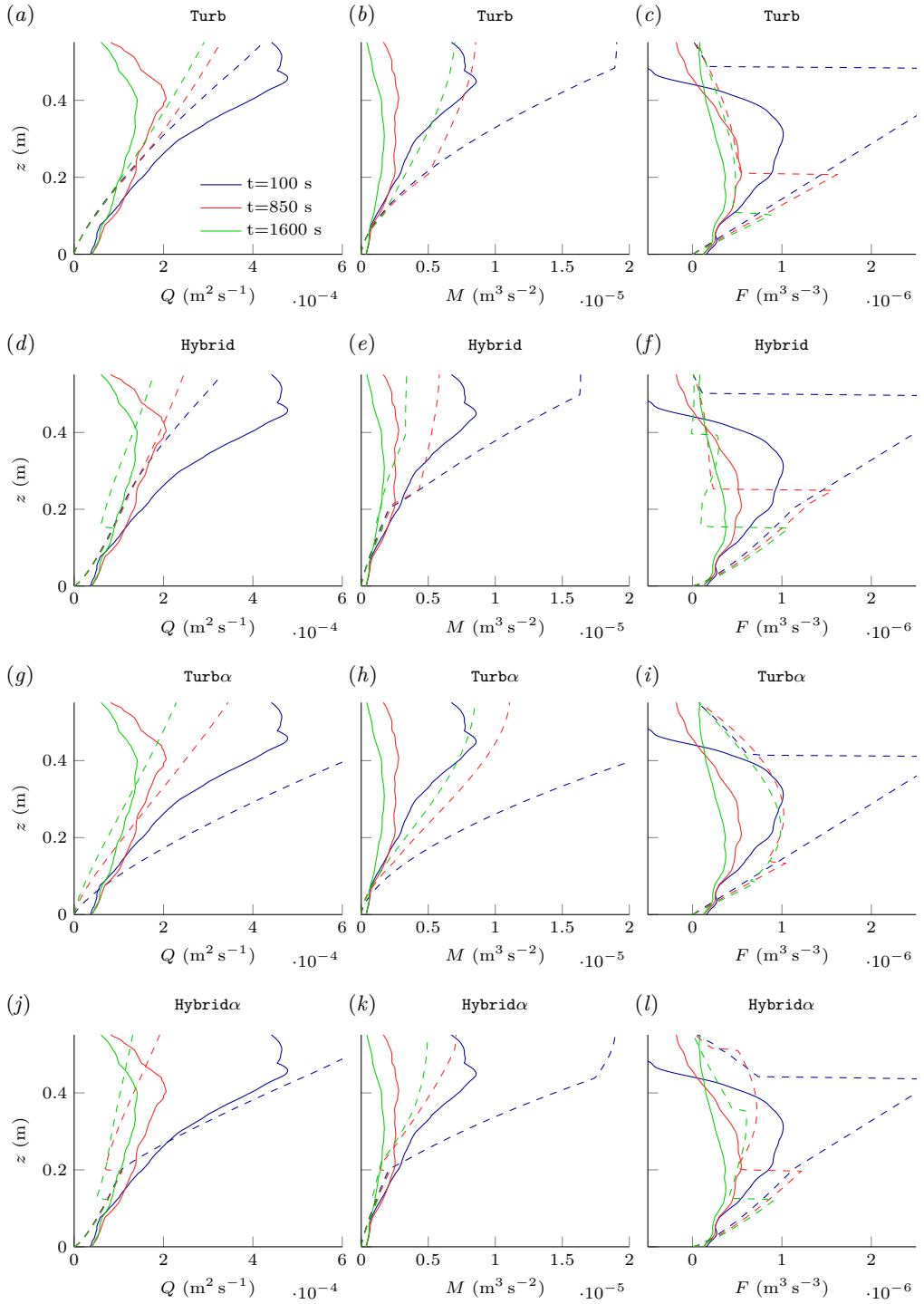


Figure 13: Buoyancy flux F , volume flux Q and momentum flux M in the plume, with experimental results (continuous lines) and models predictions (dashed lines).

ambient fluid reduces the entrainment coefficient to the extent that the entrainment coefficient of the plume is found to reduce by a factor of 10. Furthermore we observe the presence of a laminar flow adjacent to the heated wall which causes a transport of heat to the top that is almost without mixing. This results in the formation of a hot layer near the top boundary, which hardly mixes and persists throughout the flow evolution. As a consequence, the ambient stratification has typically top, middle and bottom regions referred to as (A), (B) and (C), respectively, with each having a different evolution in the temperature profile as shown in figure 5. These observations, as well as the detailed comparison with a simple theoretical MTT model, have motivated us to improve the modelling.

To predict the plume characteristics and the ambient stratification, the initial modelling (**Turb**) is based on the original Morton *et al.* (1956) entrainment model, here adapted to a wall with constant temperature. This model has been enriched both by a variable entrainment coefficient α and by differentiating a laminar and a turbulent zone. The treatment of the laminar zone has been tackled using the framework proposed by Worster & Leitch (1985). At the critical height of the transition between the laminar and the turbulent region, the values calculated from this laminar model are implemented as lower boundary conditions into the turbulent model. This leads to a new hybrid model in which we consider successively an average value (**Hybrid**) and an exponential fit of the experimentally measured values in time (**Turb α**) for the entrainment coefficient (α). Thus four different model results are obtained.

For the present experiments in the range of $Ra_H = 5.4 \times 10^{10}$ and $Pr = 5.5$, detailed comparison with the experimental results show that the **Turb** model describes to leading order the evolution of the wall plume and interior stratification, unlike the transitional character of the wall plume. This model, as well as the experimental measurement results, is very close to the results reported by Cooper & Hunt (2010) for a constant buoyancy flux modelled with saline water across the wall. In view of both the Schmidt number being 100 times larger than the present Prandtl number and the difference in forcing, this result was not evident in advance.

Better model results are achieved when taking into account the laminar onset of the plume with the **Hybrid** model, and a significant improvement is found when the variation of α is implemented (**Turb α** model). Indeed, our results have brought experimental support for an entrainment coefficient which varies by an order of magnitude due to the gradual stratification in the environment. The decrease of α in time is determined empirically from the measurements and corresponds well with the modelling detailed in appendix B that is based on the global entrainment coefficient of both the homogeneous and the stratified layers. For further modelling it is possible to use the expression for α (B3) for two given entrainment coefficients α^h and α^s and to employ the model values of w and z_{fr} . The drawback is that the value of w is not very accurately predicted (figure 13(e) and (k)) as mentioned above, presumably due to the absence of the shear stress in the modelling. Furthermore, care should be taken with the entrainment value α^s which depends on stratification and plume strength, and may therefore vary from one experiment to another.

Since the laminar-turbulent flow transition depends on the critical Grashof number, proportional to H^3 , the relative part of the laminar boundary layer changes strongly with a small increase in height. For smaller heights than used here, it is clear that the **Hybrid** model should be used instead of the **Turb** model. Of course in the limit of a very high reservoir with a configuration analogous to the present one, the laminar layer would represent a fraction of the total height and we would then expect the influence of using the **Hybrid** model to be limited. However, in this case a better modelling may

nevertheless be achieved because of the qualitative changes it implies on the boundary condition for small z . Because of its variation with background stratification, the use of a variable entrainment coefficient remains important for a precise modelling in all cases.

The T-LIF technique has been shown to be an accurate and efficient method to measure the temperature in a plane, with a precision of 0.2°C . It allows density fields to be acquired non-intrusively next to the velocity field. This is of particular interest for convection experiments where both are strongly related. The high resolution allows details of small structures as well as large-scale circulation to be obtained. Here, we have applied this method to a plume along a sidewall and have been able to get instant views of the plume establishment and to obtain precise temperature profiles to describe stratification. This experimental technique is ready to produce benchmark data to numerical simulations, and further experiments need to be conducted to exploit the method at its full potential.

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Appendix A. Heat transfer at the wall

Equation (4.3) expresses the heat flux at the wall per unit surface, φ , as a function of the temperature at the wall, T_w , and the temperature in the ambient, T_e . Based on the conservation of the heat flux φ through the wall, §4.3 explains how the buoyancy at the wall is modelled as a function of the temperature in the side compartment. Here it is shown how equation (4.32) is obtained.

Suppose \bar{U}_c is the averaged fluid velocity in the heated compartment, e_c the width of this compartment and e the width of the aluminium plate which separates the compartment from the experimental tank (figure 14(a)). Three types of heat transfers coexist, i.e. the forced convection within the compartment, the conduction inside the aluminium plate, and the natural convection in the tank. As the heat flux is conserved one can write

$$\varphi = \frac{T_c - T_{cw}}{1/h_c} = \frac{T_{cw} - T_w}{e/\lambda_a} = \frac{T_w - T_e}{1/h_w}, \quad (\text{A } 1)$$

where λ_a is the thermal conductivity of aluminium and h_c is the heat transfer coefficient characterising exchanges between the heated fluid in the side compartment and the adjacent surface of the aluminium plate. Rewriting φ in terms of known temperatures this yields

$$\varphi = \frac{T_c - T_e}{\frac{1}{h_c} + \frac{e}{\lambda_a} + \frac{1}{h_w}}. \quad (\text{A } 2)$$

Thus, the global heat transfer coefficient which has been introduced in §4.3 can be written as $h_g = 1 / \left(\frac{1}{h_c} + \frac{e}{\lambda_a} + \frac{1}{h_w} \right)$.

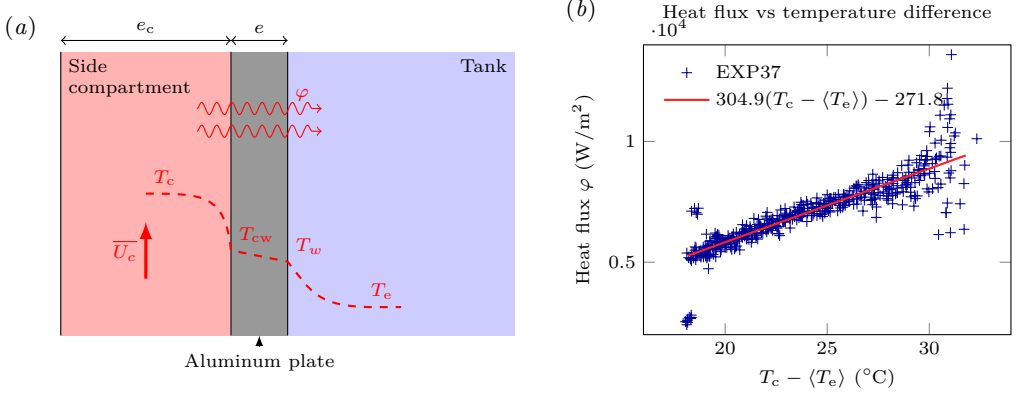


Figure 14: (a) Schematic view of the heat transfer between the heated compartment and the experimental tank. T_c is the temperature in the compartment (imposed), T_e is the temperature in the tank at a distance from the wall, and T_{cw} and T_w are intermediate wall temperatures. The dashed line sketches a temperature profile along a horizontal line. (b) Experimental determination of coefficient h_g averaged over the height. The instant thermal flux is plotted as a function of the temperature difference between the side compartment and the averaged ambient fluid. The global averaged heat transfer coefficient for the wall $\langle h_g \rangle$ is then given by the coefficient of the linear regression.

A.1. Free convection in the tank

Using the expression of h_w given in (4.4), one can rewrite the heat flux as

$$\varphi = K_w k \left(\frac{g\beta}{\nu\kappa} \right)^{\frac{1}{3}} (T_w - T_e)^{\frac{4}{3}} = h_g (T_c - T_e), \quad (\text{A } 3)$$

which gives more conveniently

$$C_1 (y + x)^{\frac{4}{3}} = h_g (1 + x) \quad (\text{A } 4)$$

with

$$x = \frac{T_1 - T_e}{T_c - T_1}, \quad y = \frac{T_w - T_1}{T_c - T_1}, \quad C_1 = K_w k \left(\frac{g\beta}{\nu\kappa} \right)^{\frac{1}{3}} (T_c - T_1)^{\frac{1}{3}}. \quad (\text{A } 5)$$

Then, using Taylor series for x close to a value x_0 , to be determined yields

$$y = \left[\frac{3}{4} \left(\frac{h_g}{C_1} \right)^{\frac{3}{4}} (1 + x_0)^{-\frac{1}{4}} - 1 \right] x + \left(\frac{h_g}{C_1} \right)^{\frac{3}{4}} \left[(1 + x_0)^{\frac{3}{4}} - \frac{3}{4} \frac{x_0}{(1 + x_0)^{\frac{1}{4}}} \right] + o((x - x_0)^2). \quad (\text{A } 6)$$

The average temperature of the ambient over the entire duration of the experiment is $\langle T_e \rangle = 29^\circ\text{C}$. This temperature is then substituted in the expression of x to obtain a convenient value for x_0 . We will assume that the variations of h_g with height are negligible. Then, we can write $\langle h_g \rangle = h_g$, and estimate this value empirically from the instant power injected into the tank, as obtained from the average temperature difference between that in the compartment T_c , and that in the tank T_e . Figure 14(b) shows we obtain $\langle h_g \rangle \approx 300 \text{ W m}^{-2} \text{ K}^{-1}$.

Furthermore, we assume that the temperature inside the plate (T_w) is homogeneous because T_c is uniform and the thermal diffusion within the aluminium is 1000 times higher than in water. With $T_c = 53.6^\circ\text{C}$ and T_1 taken as the initial temperature (21.55°C), one

obtains

$$y = -0.567x + 0.544, \quad (\text{A } 7)$$

so that this reads for the buoyancy

$$\delta_w = -0.567\langle\delta_e\rangle + 0.544\delta_c, \quad (\text{A } 8)$$

where the averaged value of δ_e is used since δ_c and δ_w are assumed to be uniform with z .

Appendix B. Details about the entrainment coefficient

In order to calculate α from the data, we use a linear regression based on the least squares method between $\frac{d(bw)}{dz}$ and w . Given a set of N experimental points for which we know the characteristic velocity w_i and the spatial derivative of the volume flux, $\frac{d(bw)}{dz}_i$, corresponding to a horizontal plume slice of height Δz_i , we evaluate the entrainment coefficient as

$$\alpha = \frac{\sum_{i=1}^N \Delta z_i w_i \frac{d(bw)}{dz}_i}{\sum_{i=1}^N \Delta z_i w_i^2}. \quad (\text{B } 1)$$

For convergence, at least 40 experimental points are needed.

The turbulent domain is split into two slices, a ‘homogeneous’ zone denoted ‘h’ corresponding to $z_{\text{crit}} \leq z < z_{\text{fr}}$ and a ‘stratified’ zone denoted ‘s’ where $z_{\text{fr}} \leq z < H$ (see figure 15(a)). We denote the entrainment coefficient of these two zones as α^h and α^s . In the homogeneous zone α^h is constant. In the stratified zone, the entrainment rate α^s remains approximately constant (see figure 10(b) for $t > 1200$ s) which is coherent with the almost constant temperature gradient (see figure 5) and the weak variation in the plume intensity (less than a factor two). Based on equation (B 1) α^h and α^s can be expressed as

$$\alpha^h = \frac{\sum_{i=i_{\text{crit}}}^{i_{\text{fr}}} \Delta z_i w_i \frac{d(bw)}{dz}_i}{\sum_{i=i_{\text{crit}}}^{i_{\text{fr}}} \Delta z_i w_i^2} \quad \text{and} \quad \alpha^s = \frac{\sum_{i=i_{\text{fr}}+1}^{i_H} \Delta z_i w_i \frac{d(bw)}{dz}_i}{\sum_{i=i_{\text{fr}}+1}^{i_H} \Delta z_i w_i^2}, \quad (\text{B } 2)$$

where i_{crit} , i_{fr} and i_H are the indices of z_{crit} , z_{fr} and H , respectively. Then one can express the global entrainment coefficient on the turbulent zone as

$$\alpha = \frac{\alpha^h(z_{\text{fr}} - z_{\text{crit}})\langle w_i^2 \rangle^h + \alpha^s(H - z_{\text{fr}})\langle w_i^2 \rangle^s}{(H - z_{\text{crit}})\langle w_i^2 \rangle^{h+s}}, \quad (\text{B } 3)$$

where $\langle \cdot \rangle^x$ denotes an average over the “x” region.

In accordance with the turbulent curve plotted in figure 10(b), the entrainment values $\alpha^h = 0.072$ and $\alpha^s = 0.0038$ are introduced. Using experimental values for $w(z)$ and the front position z_{fr} , the variation in the turbulent entrainment coefficient is calculated (see figure 15(b)).

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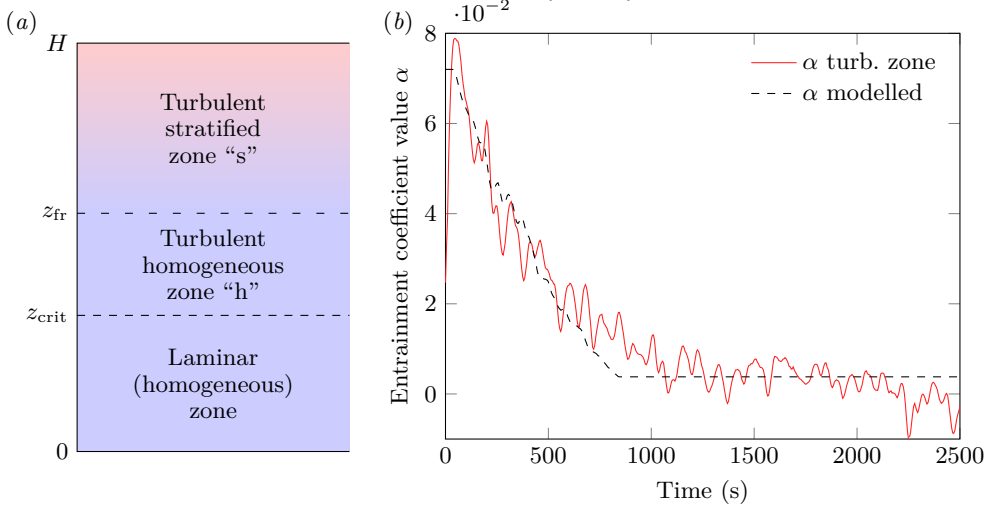


Figure 15: (a) Scheme of the computation zones and the associated variables. (b) Comparison between α computed over the entire turbulent zone (red) and α modelled using equation (B 3) (dashed black).

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