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Experimental study of the hydrodynamic expansion following a nanosecond repetitively pulsed discharge in air

Da A. Xu,^{1,2,a)} Deanna A. Lacoste,^{1,2} Diane L. Rusterholtz,^{1,2} Paul-Quentin Elias,³ Gabi D. Stancu,^{1,2} and Christophe O. Laux^{1,2}

¹Laboratoire EM2C, CNRS/UPR 288, Grande Voie des Vignes, Chatenay-Malabry 92295, France

²Ecole Centrale Paris, Grande Voie des Vignes, Chatenay-Malabry 92295, France

³ONERA DMPH, Chemin de la Hunière, Palaiseau 91761, France

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We report on an experimental study of the hydrodynamic expansion following a nanosecond repetitively pulsed (NRP) discharge in atmospheric pressure air preheated up to 1000 K. Single-shot schlieren images starting from 50 ns after the discharge were recorded to show the shock-wave propagation and the expansion of the heated gas channel. The temporal evolution of the gas temperature behind the shock-front is estimated from the measured shock-wave velocity by using the Rankine-Hugoniot relationships. The results show that a gas temperature increase of up to 1100 K can be observed 50 ns after the nanosecond pulse. © 2011 American Institute of Physics. [doi:10.1063/1.3641413]

Nanosecond repetitively pulsed (NRP) discharges are being increasingly used in various applications, in particular in plasma-assisted combustion,^{1–4} nanomaterials synthesis,⁵ x-ray generation,⁶ biomedicine and biochemistry,^{7,8} and aerodynamic flow control.⁹ The key advantage of NRP discharges is their high energy efficiency, and thus low power consumption, to produce highly reactive atmospheric pressure air plasmas. For instance, NRP sparks can dissociate a large fraction of oxygen molecules and heat the gas by about 1000 K within a few tens of nanoseconds following each pulse.^{10–13} This effect has been attributed to the so-called ultrafast mechanism initially proposed by Popov.¹⁴ The ultrafast mechanism corresponds to the following two-step process:

In the first step, which takes place during the high voltage pulse, nitrogen molecules are excited by electron impact to electronic states such as $A^3\Sigma$, $B^3\Pi$, and $C^3\Pi$. Next, these excited electronic states of N_2 undergo a quenching process by oxygen molecules,¹⁵ producing atomic oxygen radicals and heat release within a few tens of nanoseconds. This ultrafast production of O was experimentally verified.¹⁶ The ultrafast heating was observed, however, with the assumption that the rotational and translational modes of N_2 are equilibrated at all times.^{12,13}

Direct evidence to further confirm the ultrafast heating by NRP discharges may be obtained by imaging the shock-wave propagation starting from a few nanoseconds after the end of the NRP discharge. Most authors^{17–21} investigated hydrodynamics of spark in air or fuel-air mixtures with longer duration ($>0.5 \mu\text{s}$) and greater energy input ($>10 \text{ mJ}$), in

a time region from several microseconds following the spark. Reinmann and Akram²² have extended the investigated time region to as early as 200 ns following the spark. To date, however, the hydrodynamics of the NRP sparks have not been examined. The small plasma dimensions, the short lifetime (tens to hundreds of nanoseconds), and the weakness of the shock-waves produced by the NRP discharges in heated air, make the investigation challenging.

In this letter, we present time-resolved, single-shot schlieren imaging of shock-wave and heated gas channel expansion induced by the NRP discharges. The NRP discharge setup^{12,13} is sketched in Fig. 1. The NRP discharges were generated between two refractory steel pin electrodes separated by an adjustable gap distance of 2–4 mm. The pins had a diameter of 2 mm with sharpened tips and were placed in a vertical pin-pin configuration with the anode above the grounded cathode. The NRP discharges were created in preheated (300–1000 K) atmospheric pressure air. The room-temperature or preheated air flow was introduced from the cathode side, flowing parallel to the pins at $0.5\text{--}2.6 \text{ ms}^{-1}$. High voltage pulses of 10 ns duration, 1–10 kHz pulse repetition frequencies (PRF), and 5–10 kV amplitude were produced by a solid-state pulse generator (FID). The voltage across the electrodes was measured with a LeCroy PPE20kV high-voltage probe. The current was measured with a

FIG. 1. (Color online) Schematic of the NRP discharge setup.

^{a)}Author to whom correspondence should be addressed. Electronic mail: da.xu@ecp.fr.

Pearson coil (model 6585). Both signals were recorded simultaneously with a 1 GHz oscilloscope. The spark energy was deduced from the voltage and current waveforms.¹³

The schlieren illumination was provided by a Nanolite flash lamp, which produces high-intensity, short duration arcs in air with a pulsing frequency of 1 Hz and a flash duration of about 20 ns, sufficient to freeze the shock-wave for single-shot image recording. An intensified charge-coupled device (ICCD) camera (Princeton Instruments PI-MAX) with a 512×512 pixel resolution recorded the schlieren images with a well-defined exposure time window. The sensitivity of the schlieren system was ensured by a 2-m focal length spherical mirror. The synchronization of the NRP discharge with the flash lamp and the ICCD camera was achieved with a Berkeley Nucleonics (BNC 575) multi-channel pulse delay generator (200 ps RMS jitter).

Measurements were performed over a range of conditions. Here, three cases will be presented (see Table I). The first two cases were investigated in a 300 K air flow with a gap distance of 2 mm, and Case 3 was in an air flow pre-heated to 1000 K, with a 4 mm gap distance, similar to the discharges studied by Pai *et al.*¹² and Stancu *et al.*¹⁶ The reduced electric field is determined to estimate fraction of energy transferred into heating.²³ An example of successive schlieren images is shown in Fig. 2. The temporal evolutions of the shock-wave radius and of the heated gas channel radius in the middle plane of the gap are measured from the schlieren images and are shown in Fig. 3. The shock-wave velocity can then be deduced from the measured shock-wave radius evolution. It is found that initially the shock-wave accompanies the heated gas channel growth in a cylindrical shape and then propagates beyond the heated gas channel to form a spherical wave. The separation time can be varied from about 250 to 500 ns depending on the configuration case studied. As it propagates, the shock-wave degenerates into a sound wave. In all cases, the heated gas channels grow from a radius of about 0.4 mm and eventually dissipate in the surrounding air.

There are two major findings in the analysis of the results. First, the surrounding air temperature is inferred from the velocity of the pressure-wave degenerated from the discharge-induced shock-wave. The velocity of pressure-wave lies so close to the speed of sound that the pressure-wave can be considered as a sound wave. Sound-wave velocities in cases 1 and 2 are found to be 346 ± 20 m/s and 335 ± 20 m/s, respectively, i.e., close to the value of the speed of sound in air at 300 K (347 m/s), indicating that in those cases, the NRP discharges do not appreciably heat the surrounding air. In contrast, in case 3, the sound-wave velocity is found to be 773 ± 50 m/s, corresponding to the sound speed at the temperature of 1500 ± 200 K, showing that the

Case 1 (Gap distance = 2 mm, PRF = 1 kHz, $T_{\text{air flow}} = 300$ K)

Case 2 (Gap distance = 2 mm, PRF = 10 kHz, $T_{\text{air flow}} = 300$ K)

Case 3 (Gap distance = 4 mm, PRF = 10 kHz, $T_{\text{air flow}} = 1000$ K)

FIG. 2. Examples of single-shot successive schlieren images. The exposure time is 20 ns. Numbers below the images represent the elapsed time after discharge initiation. The minus sign indicates a time before the discharge. Note, in Case 2, that the gas heated by previous discharges has not been fully advected, but that the shock-wave expands at the same velocity as in Case 1.

NRP discharges in this case heat the air in the vicinity of the discharge to an initial (i.e., preceding each pulse) temperature of about 1500 K, a value consistent with the temperature

TABLE I. Summary of the three cases investigated.

	Gap distance	Air flow temperature	PRF	Electric energy per pulse	E/N	Energy transferred into heating ^a	T_{initial}	ΔT_{max}
Case 1	2 mm	300 K	1 kHz	0.15 mJ	150 Td	0.03 mJ	~ 300 K	< 200 K
Case 2	2 mm	300 K	10 kHz	0.5 mJ	126 Td	0.06 mJ	~ 300 K	< 200 K
Case 3	4 mm	1000 K	10 kHz	1.6 mJ	413 Td	0.57 mJ	~ 1500 K	1100 K

^aFraction of electric energy transferred into heating, based on the calculation results of Alexandrov *et al.*²³

FIG. 3. Temporal growth of the shock-wave and heated-gas-channel radius measured in the middle plane of the gap for Cases 1 and 3. The data for Case 2 are not plotted because they are very similar to those of Case 1.

measurement based on $N_2(C-B)$ spectra of the same type of discharges.^{12,13} The differences between the three cases can be explained by the different fraction of electric energy transferred into heating, as shown in Table I.

Second, the temperature and pressure behind the shock front can be obtained from the Rankine-Hugoniot relationships

$$\frac{p_2}{p_1} = \frac{2\gamma_1}{\gamma_1 + 1} \left(\frac{u_s}{u_1} \right)^2 - \left(\frac{\gamma_1 - 1}{\gamma_1 + 1} \right), \quad (1)$$

$$\frac{T_2}{T_1} = \left(\frac{p_2}{p_1} \right) \cdot \frac{(\gamma_1 + 1) + (\gamma_1 - 1)(p_2/p_1)}{(\gamma_1 - 1) + (\gamma_1 + 1)(p_2/p_1)}, \quad (2)$$

where s denotes the shock-wave, 1 the condition ahead of the shock-wave (initial condition), 2 the condition behind the shock-wave, u_1 the acoustic velocity, and γ_1 the ratio of specific heats. From these relations, with the values of shock-wave velocity deduced from Fig. 3 and assuming $T_1 = 300$ K in cases 1 and 2, $T_1 = 1500$ K in Case 3, the temperatures evolution behind the shock-front are found (see Fig. 4).

Fig. 4 shows that in cases 1 and 2, the temperature behind the shock-wave remains close to 300 K. In contrast, in case 3, the maximum temperature reaches 2600 K in the first 100 ns and decreases rapidly afterwards. After the peak, the shock-wave experiences a deceleration and then again an acceleration between 260 and 500 ns (which is caused by the secondary discharge fired by the electrical circuit at about 250 ns), and the temperature increases to a second peak of about 2300 K.

The present results confirm the ultrafast heating mechanism of NRP discharges. This mechanism is consistent with the two-step mechanism involving electron-impact excitation of N_2 followed by the dissociative quenching of the excited electronic states of N_2 by O_2 . The ultrafast heating may have important applications in aerodynamic flow control, gas treatment, plasma-assisted combustion, or nanomaterial fabrication.

FIG. 4. Temperature behind the shock-wave versus time, based on Rankine-Hugoniot relations.

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