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Investigation of Dynamic Crack Propagation and Arrest for Pulse Loaded specimens Made from a Modified MoV-Steel (KS22) by Means of a Hopkinson-Pressure-Bar

K. Kussmaul and U. Mayer

Staatliche Materialprüfungsanstalt (MPA) University of Stuttgart, Pfaffenwaldring 32, 70569 Stuttgart, Germany

Abstract. Experimental and numerical investigations on crack initiation under transient loading, crack propagation behaviour and crack arrest of 17 MoV 84 (mod.) using fracture mechanics methods are described. Results of Split-Hopkinson-bar-tests using a single-specimen method and a high-speed camera are compared with multi specimen results and values obtained from other specimens and load velocities.

Résumé. Ce rapport décrit des études expérimentales et numériques concernant l'initiation des fissures sous charge transitoire, ainsi que leur propagation et arrêt dans le matériau 17 MoV 84 (mod.) en utilisant des méthodes de mécanique de fracture. Les résultats des essais Hopkinson avec une méthode à une éprouvette et une caméra à action rapide ont été comparés à ceux obtenus avec plusieurs éprouvettes et aux valeurs d'autres éprouvettes et d'autres vitesses de charge.

1 Introduction

The deformation and fracture behaviour of steels under high loading rates is quite different from that under quasistatic loading.

It is known that there is an increase in the yield stress as well as a decrease in fracture toughness of ferritic and ferritic-perlitic steels under increasing loading rate [1, 2]. These effects occur particularly under plastic strain rate $\dot{\epsilon}_{pl} > 10^3 \text{ s}^{-1}$ as they may be found in the plastic zone and the process zone of dynamically loaded cracks. An averaged change rate \dot{K} of the stress intensity factor or the averaged change rate \dot{J}_{pl} of the plastic part of the *J-Integral* are used as typical velocity parameter.

A large number of methods is available in the load rate range $\dot{K} > 10^6 \text{ MPa}\sqrt{\text{m}}$. A crack is loaded by transient elastic stress fields generated by mechanical shock or explosive load or electro-magnetic forces. Basically one can distinguish between two methods: direct loading of comparably small specimens by a shock [3] and use of a cracked specimen as an elastic wave guide [4]. The advantage of the first arrangement is that the stress pulse loading the crack can precisely be measured. This simplifies the necessary numerical calculation of the near crack field parameter. On the other hand, for the direct shock load it is generally uncertain to determine the boundary conditions necessary for the numerical analysis. To obtain a homogeneous and stable pressure pulse over the cross-section it has to propagate over a long distance. The specimen length can be kept short if a separate reusable waveguide, which is acoustically coupled to a specimen, is used for pulse generation.

The $K_{I,d}$ -determination in this method has been based on an elastodynamic numerical analysis of the test arrangement using the measured loading pulse. Using several identical specimens a critical fracture stress σ_{crit} is determined by variation of the pulse amplitude with simultaneous measuring of the crack propagation, from which $K_{I,d}$ follows

$$K_{I,d} = \max \{ K_I(t) \} = \sigma_0 \sqrt{\pi \cdot a} \cdot g(a/w, W, \text{pulse}) \quad (1)$$

$g(a/w, W, \text{pulse})$ is a numerically derived function of the normalized crack length a/W , the specimen width W and the form of the pressure pulse. A crack start can be assumed in the maximum of the $K(t)$ -

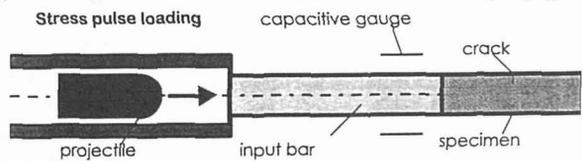


Figure 1: Hopkinson bar fracture test

curve.

The advantage of this method is the simple dynamic crack loading, which facilitates the theoretical modelling. Oscillations due to impact loading did not occur because a definite pulse is applied without reflection to the specimen. By measurement of stress in the specimen and in the input bar using strain gauges or a coaxial capacitive gauge this can be very well examined. This method has been used successfully for determining K_{Id} on various steels in the lower shelf and the transition area.

2 Material

The material KS22 of the research project Integrity of Components (FKS) [5] is a melt based on 17 MoV 8 4. Heat treatment tests were carried out on this material with the aim of obtaining a model material showing low notch impact energy in the upper shelf for isotropic material properties, high brittle fracture transition temperature and high strength. The melt was austenitized at 1050 °C and tempered at 640 °C 7 h. The upper shelf with a notch impact energy of 40 J to 50 J is obtained at approx. 300 °C. The NDT temperature was estimated according to P4-criterion at 250 °C, since it was not possible to conduct drop weight tests in this temperature range. Particular characteristics of the bainitic structure of this material are high hardness, large grains (grain size 4 according to ASTM E 112) and many non-metallic inclusions. The specimens were taken out in T- resp. TL direction as in the project "Crack initiation under impact load" [6].

3 Aim

Previous investigations have proved [7] that the material 17 MoV 8 4 (mod) is at room temperature in the lower shelf of the notch impact energy temperature curve and fails macroscopically without plastic deformation independent of the loading rate. Therefore, this material is well suited to tests for comparison of the multi-specimen method and a single specimen method (direct measuring of crack start). All investigations were carried out on smooth specimens, i.e. without side grooves and at room temperature. $K_{I_{max}}$ was calculated from the pulse amplitudes and the initial crack length using the dynamic correction functions, discussed above. This yields a fracture toughness $17.6 \text{ MPa}\sqrt{\text{m}} < K_{Id} < 20.5 \text{ MPa}\sqrt{\text{m}}$ (at $\dot{K} \approx 2 \cdot 10^6 \text{ MPa}\sqrt{\text{m}}/\text{s}$); mean value $K_{Id} = 19 \text{ MPa}\sqrt{\text{m}}$. Results of direct measuring of the crack start using the striation method and the induction probe are $K_{Id} = 18,5 \text{ MPa}\sqrt{\text{m}}$ resp. $K_{Id} = 23,9 \text{ MPa}\sqrt{\text{m}}$ [8]. The results of the crack initiation investigations on tensile stress pulse loaded SECT specimens made of 17 MoV 8 4 (mod.) appeared to be promising for investigations on crack propagation and crack arrest using a high speed camera.

4 Technique of measuring

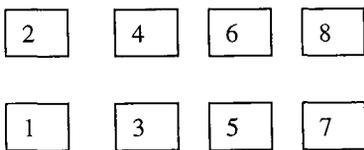


Figure 2: Arrangement of frame sequence

An electronic high-speed image converter camera IMACON 790 by Hadland Photonics was used. The electronic image converter camera converts the picture taken from the photo cathode into an electron beam which is focussed by an electrode. A frame sequence of images on a screen with phosphorescent layers is generated. A sheet film-negative is exposed by the contact method resulting in a frame sequence as presented in [Figure 2](#).

This system enables light amplifying by electronic means.

The frame frequency of the camera is determined by plug-ins with up to $2 \cdot 10^7$ frames per second (fps). Apart from discrete times of frames so-called streak-pictures are possible using the image converter camera. They consist of a one-dimensional image with continuous resolution in the nanosecond range. The measuring of dislocations and crack extensions is possible. Plug-ins with $1 \cdot 10^5$ fps, $1 \cdot 10^6$ fps and $2 \cdot 10^6$ fps and a plug-in for streak pictures with a continuous resolution of 0.1 up to 1 $\mu\text{s}/\text{mm}$ were used. Lenses for 35mm cameras can be used for the Image Converter Camera. A micro lens of a 105-mm focal length was used.

5 Experiments

Earlier investigations [9] carried out on ferritic steel in the lower shelf have shown that it was necessary to have a frame rate of 1 μ s and 0.5 μ s. First aim was to measure the crack propagation of specimens, which were fatigue pre-cracked, to a ratio of approx. $a/W=0.31$ (crack length 5.5 mm) during the first microseconds after initiation.

Using a micro-lens a 1:1 image resp. a 3-fold enlargement was focused on the luminous screen to record all frames photographically. The high contrast flash illumination of the crack border is of importance to reliable measuring. The change of surface direction in the vicinity of the crack depicts the crack. Though no clear crack could be ascertained on the surface a continuous crack extension can be seen on the photo. Microscopic measurement after tests was used to check the crack extension determined by means of high speed photography.

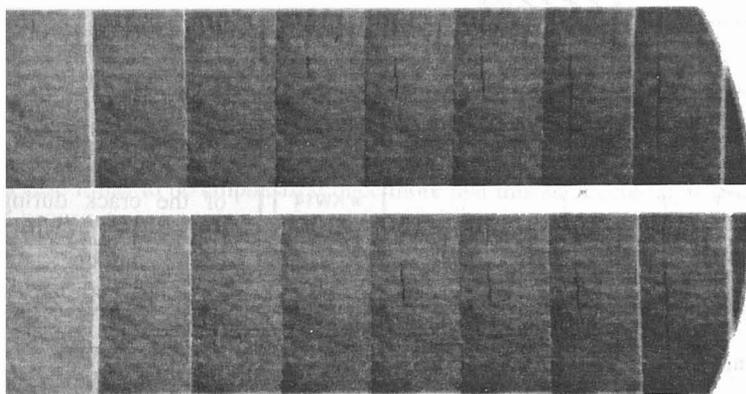


Figure 3: Frame sequence for test KW3 (100000 fps)

The first result was that no crack extension occurred during the initial tensile pulse loading of the crack at barely overcritical load. To determine the overall behaviour, a sequence with a frame rate of up to 100.000 per second was taken during several reflections of the stress pulse in the specimen. Three consecutive pulse

loadings could be observed.

Figure 3 shows pictures taken during loading of specimen KW3. The first load takes place between frame 2 and frame 3. However, no crack propagation could be found. Just after the 8th resp. 13th frame corresponding to the second or third load by tensile stress pulse a macroscopic crack propagation was detected. Following the third load no additional crack propagation occurred.

An exact numerical treatment is only possible if the crack arrest occurs following a first load of an undamaged material. Therefore the level of the load pulse was increased to such an extent that a macroscopic crack propagation of several millimetres took place.

specimen	fatigue crack	a_{tot}	total crack extension	crack extension after picture #	first picture	plug-in	t_i	t_a	K_{max}
	a/mm	mm	mm	mm(#)	t/ μ s		μ s	μ s	MPa \sqrt m
KW3	5.6	8.6	3.0	1.4(8)/3.0(13)	85.7	1.E5 fps	105.7	105.7	24.7
KW7	5.5	11.5	6.0	1.9(5)	101.5	1.E6 fps	105.5	106.5	30.5
KW8	5.6	13.1	7.5	2.69(10)	105.4	2.E6 fps	106.9	109.9	34.1
KW9	5.5	12.7	7.2	3.85(17)	102.3	2.E6 fps	104.8	110.3	34.5
KW11	5.5	15.4	9.9	6.2(9)	101.5	1.E6 fps	103.5	109.5	37.3
KW12	5.6	14.6	9.0	4.49(18)	102.3	2.E6 fps	104.3	110.8	35.4
KW13	5.4	12.9	7.5	3.72(16)	102.4	2.E6 fps	103.9	109.9	34.4
KW14	5.4	12.8	7.5	4.0(10)	102.5	1.E6 fps	104.0	111.5	34.6
KW15	5.5	16.0	10.5	7.3(14)	101.6	1.E6 fps	102.6	114.6	44.1
KW16	5.7	15.9	10.2	2.96(41.5)	91.2	streak	100.8	111.8	43.1

Figure 4: Results for crack arrest studies on 17 MoV 8 4

The results of these tests are summarised in [Figure 4](#). The initial crack length a , the crack length a_{tot} measured after the test, the crack propagation following the test and the maximum crack propagation determined by the high speed measuring are noted. Apart from the plug-in frame rate the initiation time t_i

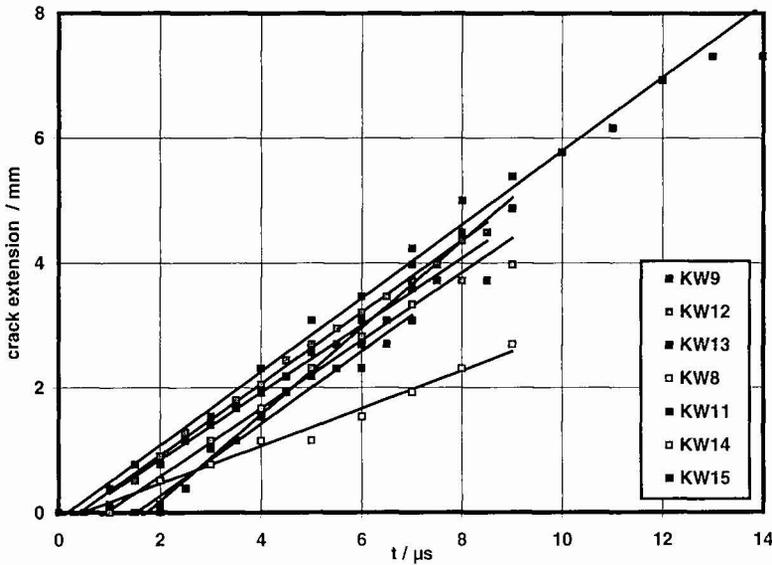


Figure 5: Crack growth for dynamic loading

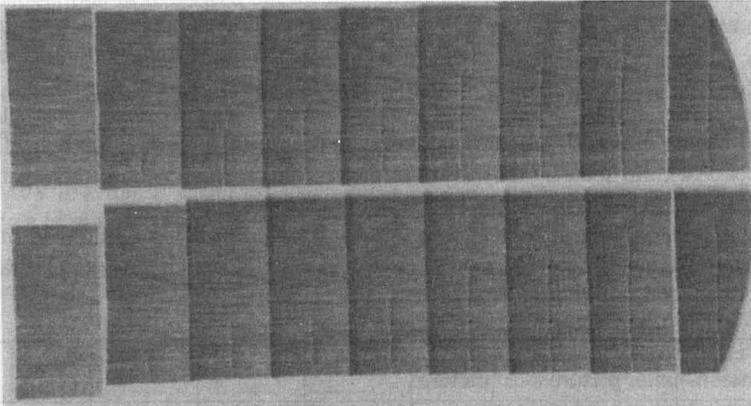


Figure 6: Frame sequence for test KW12 (2 000 000 fps)

and the arrest time t_a are given. The maximum stress intensity factor K_{max} has been calculated from the maximum of the pressure tensile stress pulse, measured on the bar using the function of a/W in equation 1.

[Figure 5](#) shows the time history of the unstable crack growth determined from the high-speed images. An almost constant velocity of the crack during the total crack growth phase can be noticed as found by other authors as well [10].

[Figure 6](#) shows the high-speed frames of test KW12. The crack propagation is evident from frame to frame. In case of a frame rate of 2 million pictures per second the duration of the crack propagation corresponds exactly to the period covered by 18 frames. Additionally, tests with 1 million frames per second were measured for verification. The total crack propagation during the period of a framing

sequence corresponds only to the crack propagation evolving during the first load. The objective of the experimental crack propagation measurement and the corresponding numerical analyses is the maximum load produced by the first tensile stress pulse. The energy stored in the specimen causes further crack propagation in the subsequent loadings. This is not to be discussed here.

Both a dynamic initiation value K_{Id} as well as a dynamically determined crack arrest value K_{IA} can be determined for the stress intensity factor. The calculation of a time history allows a single specimen test. Previously mentioned investigations found $K_{Id} = 19 \text{ MPa}\sqrt{\text{m}}$ for 17 MoV 8 4 (mod). This value was determined with the multi specimen method. The load was varied in the range, where crack propagation has just been determined. It has not been verified that the measured crack propagation developed during the first load. This distinction of results for single specimen and multi specimen methods was not necessary in earlier investigations [9] on other brittle materials, since they manifested crack propagation already in the first run of an overcritical loading pulse. The crack arrest behaviour of the

17 MoV 8 4 (mod) does not correspond with the normal one of ferritic steels in the lower shelf. Finally this fact allows the determination of dynamic crack arrest values K_{IA} for the stress intensity factor.

The streak image directs a picture of the crack at 2 mm/ μ s over the screen. Following the picture from left to right one can see the shifting crack tip on a straight line to the top. The constant crack propagation velocity is visible and confirms the results of the single images.

Probe	K_{Id}	K_{IA}	K_{max}
KW8	32.10	29.50	36.60
KW9	32.80	31.00	35.40
KW11	30.20	37.80	42.00
KW12	31.90	32.80	34.60
KW13	30.90	31.20	32.70
KW14	31.60	28.90	33.10
KW15	33.20	33.50	42.70

Figure 7: Results for crack initiation K_{Id} and crack arrest K_{IA} derived from numerical analysis

The initiation values are higher in this single specimen method than the critical values K_{Id} determined by the multi specimen method. Within the accuracy obtained K_{IA} and K_{Id} are equal. However, in this case it has to be emphasized once more that this statement applies only for the first loading pulse. Crack propagation can also be found in specimen loaded with a stress intensity factor below K_{Id} , this

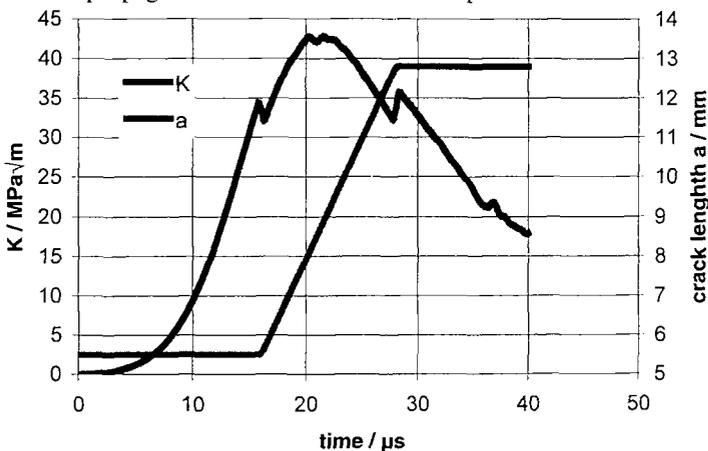


Figure 8: Stress intensity factor K and crack length a vs. time

means following loading with decreasing amplitude.

Figure 8 shows the stress intensity factor vs. time for an over-critical test (KW15) taking account of the crack propagation. Additionally the crack propagation behaviour determined from the high-speed images is shown. The transition from the propagating to the arrested crack results in a little oscillation in the numerical calculation. Within this accuracy the stress intensity factor is both at crack start and at crack arrest 32 MPa \sqrt m. Specimen which were loaded less showed crack

propagation following the test but not after the run of the first stress pulse. Since the load is lower in the following stress pulse than in the first one, local crack propagation resp. damages must have already occurred due to smaller stress without generating a global crack propagation. The shape of the crack propagation front is for the 17 MoV 8 4 (mod.) parallel to the fatigue crack so that the crack propagation measured on the surface of the specimen cannot be distinguished from the average crack propagation as ascertained in other materials [9].

Conclusions

1. Material 17 MoV 84 (mod.) manifests a behaviour deviating from the normal brittle fracture under pulse loading which is understandable because of the local inhomogeneity. Therefore, a distinction has to be made between values determined by the multi specimen method and the single specimen method.
2. This material is suitable for investigations on the running crack because there is less difference between specimen surface and specimen centre than for other brittle materials.
3. Crack velocities of approx. 500 m/s do not result in a deviation between characteristic values for dynamic crack initiation and crack arrest.

4. K_{IA} of the investigated 17 MoV 8 4 (mod.) at room temperature is higher than those K_{Ic} values determined on the compact tension specimens in quasistatic tests. This applies to higher temperatures as well. Dynamic calculations [12] made for crack arrest investigations on large-scale specimen lead to K_{IA} -values that exceed the K_{Ic} scatter band, too.

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